



Jet Propulsion Laboratory
California Institute of Technology

The SunRISE Mission

Sun Radio Interferometer Space Experiment

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California Institute of Technology)

With help from:

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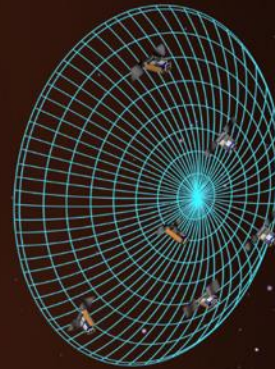
T. Joseph W. Lazio (Jet Propulsion Laboratory, California Institute of Technology)

James P. Lux (Jet Propulsion Laboratory, California Institute of Technology)

Cate Heneghan (Jet Propulsion Laboratory, California Institute of Technology)

Tim Neilsen (Space Dynamics Laboratory)

And many, many others!

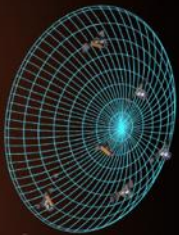


The cost information contained in this document is of a budgetary and planning nature and is intended for informational purposes only. It does not constitute a commitment on the part of JPL and/or Caltech.



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SUN RADIO INTERFEROMETER SPACE EXPERIMENT



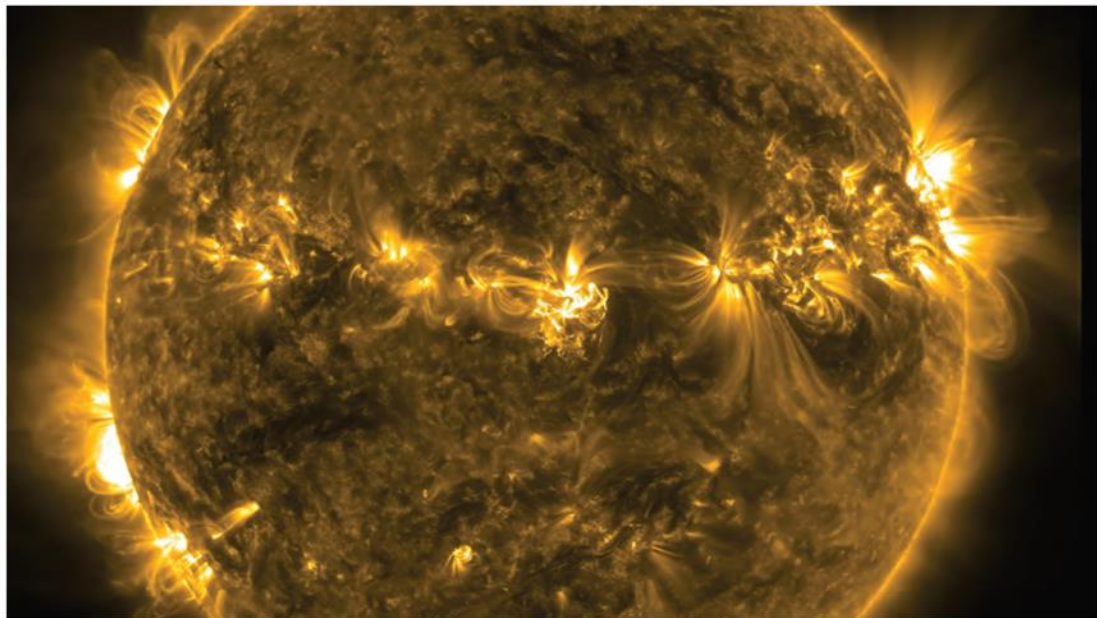
Heliophysics Decadal Survey (2013)



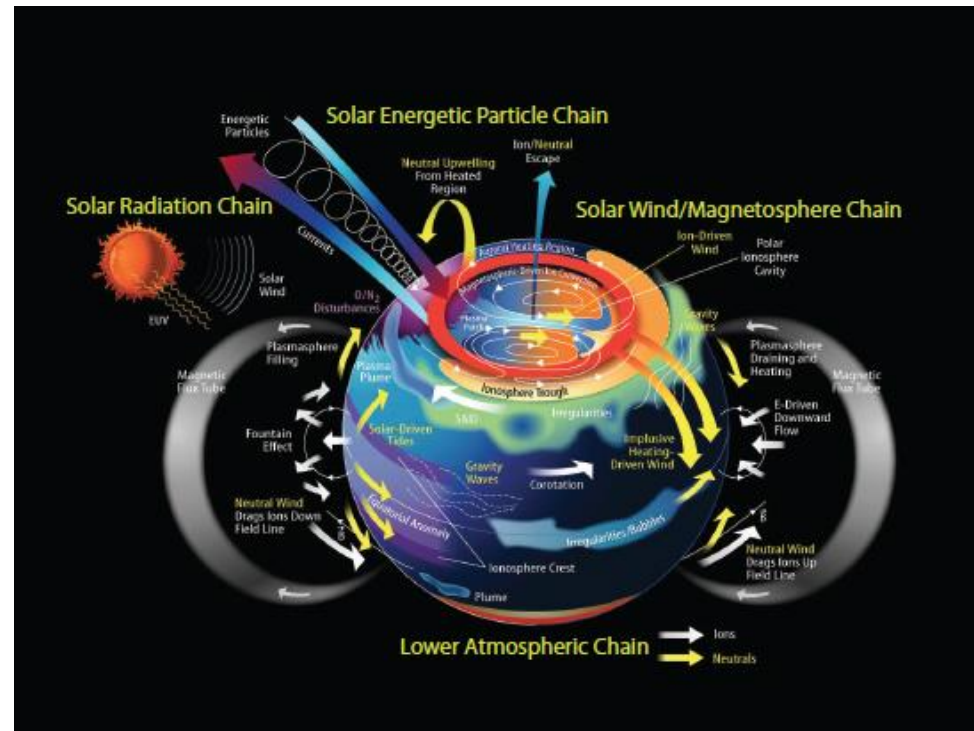
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Key Science Goal 1. Determine the origins of the Sun's activity and predict the variations in the space environment.

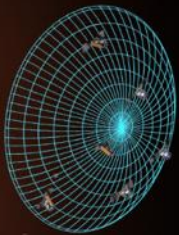
Key Science Goal 2. Determine the dynamics and coupling of Earth's magnetosphere, ionosphere, and atmosphere and their response to solar and terrestrial inputs.



Source: 2013 Heliophysics Decadal Survey



Source: 2013 Heliophysics Decadal Survey



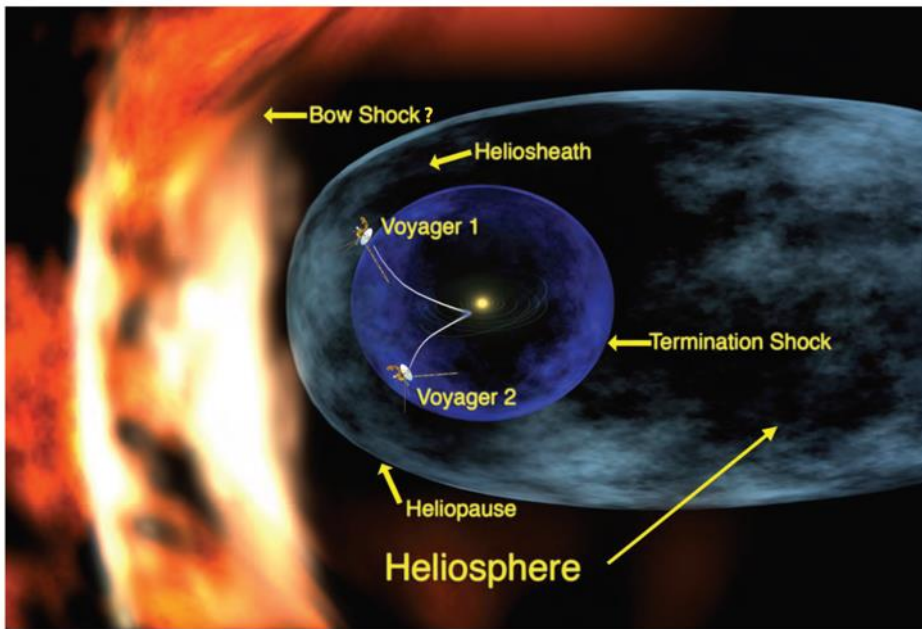
Heliophysics Decadal Survey (2013)



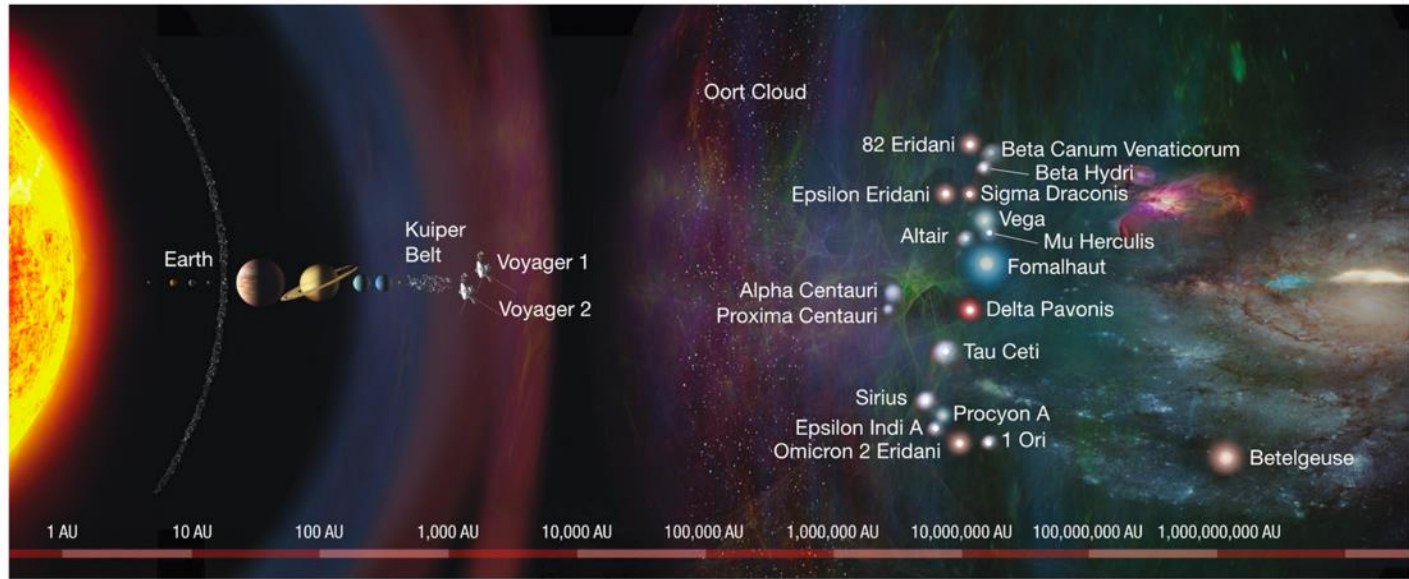
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Key Science Goal 3. Determine the interaction of the Sun with the solar system and the interstellar medium.

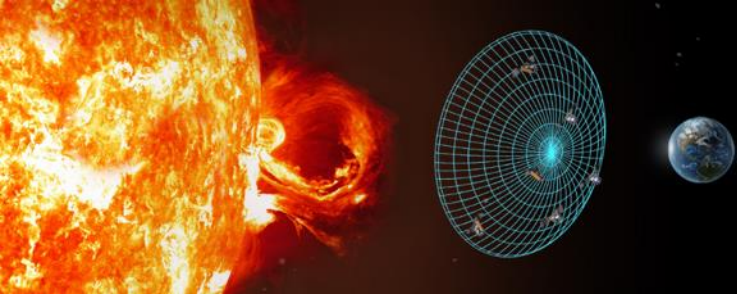
Key Science Goal 4. Discover and characterize fundamental processes that occur both within the heliosphere and throughout the universe.



Source: 2013 Heliophysics Decadal Survey



Source: Starshade Rendezvous Probe Report (2019)

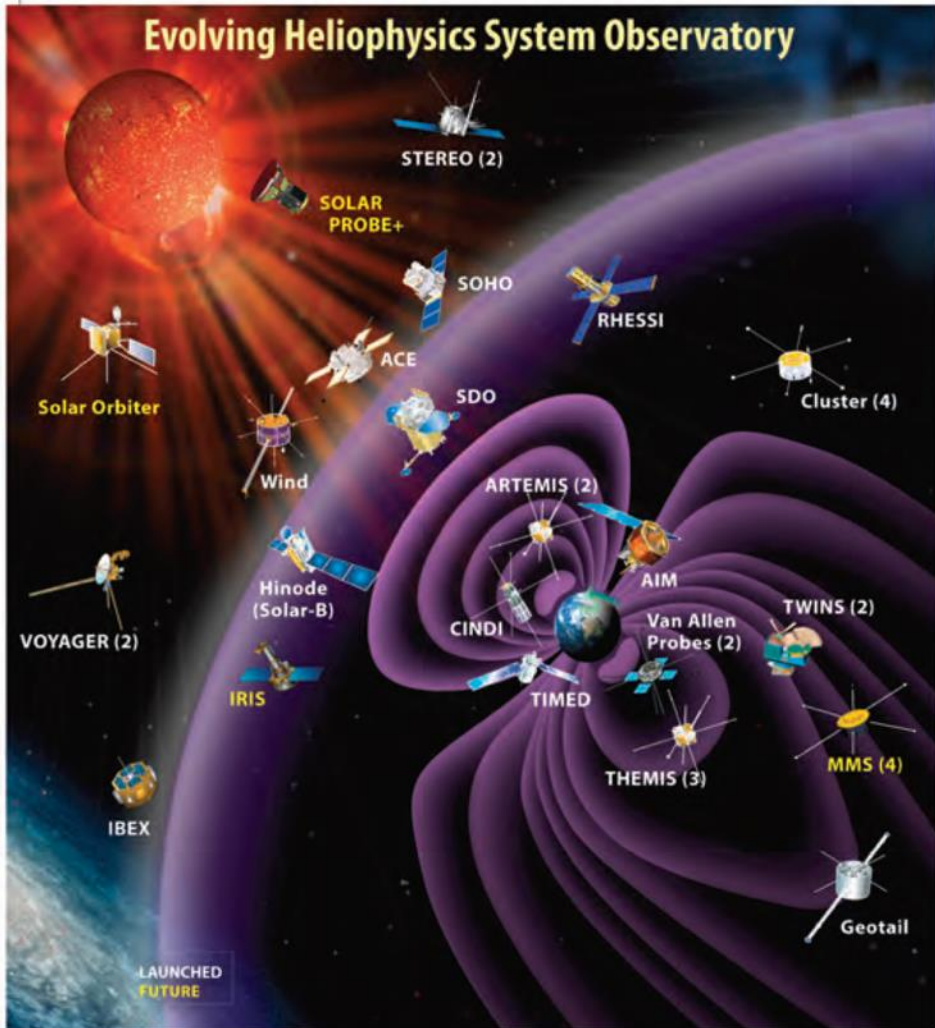


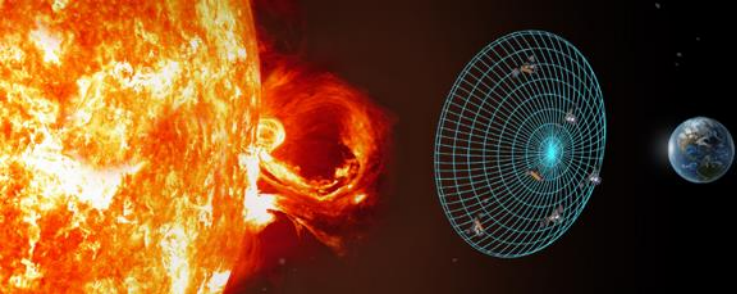
Heliophysics Observatories



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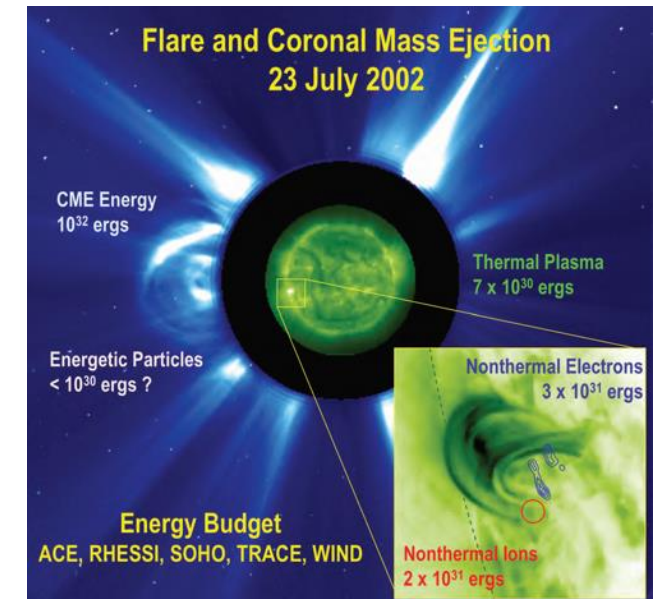
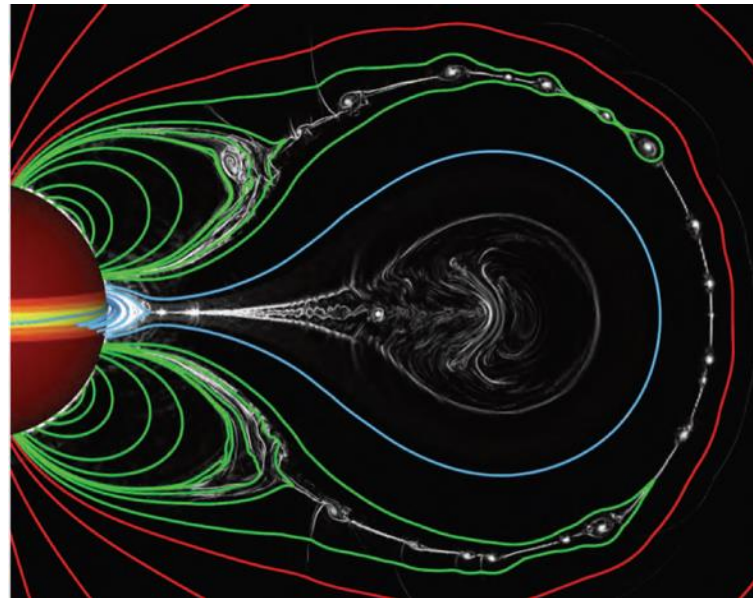
Goals

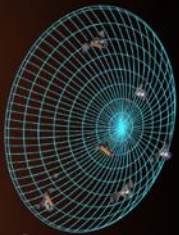


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The Sun and Heliosphere

- SHP-1 Understand how the Sun generates the quasi-cyclical magnetic field that extends throughout the heliosphere.
- SHP-2 Determine how the Sun's magnetism creates its hot, dynamic atmosphere.
- SHP-3 Determine how magnetic energy is stored and explosively released and how the resultant disturbances propagate through the heliosphere.
- SHP-4 Discover how the Sun interacts with the local interstellar medium.

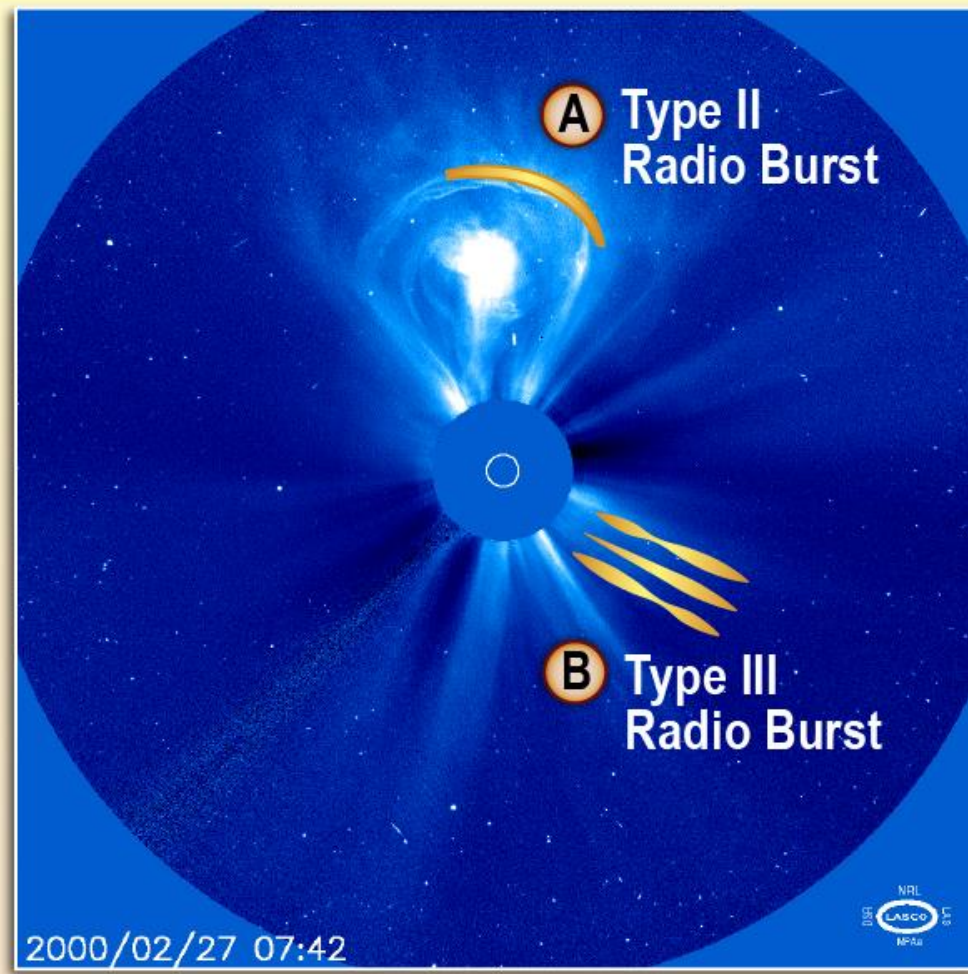
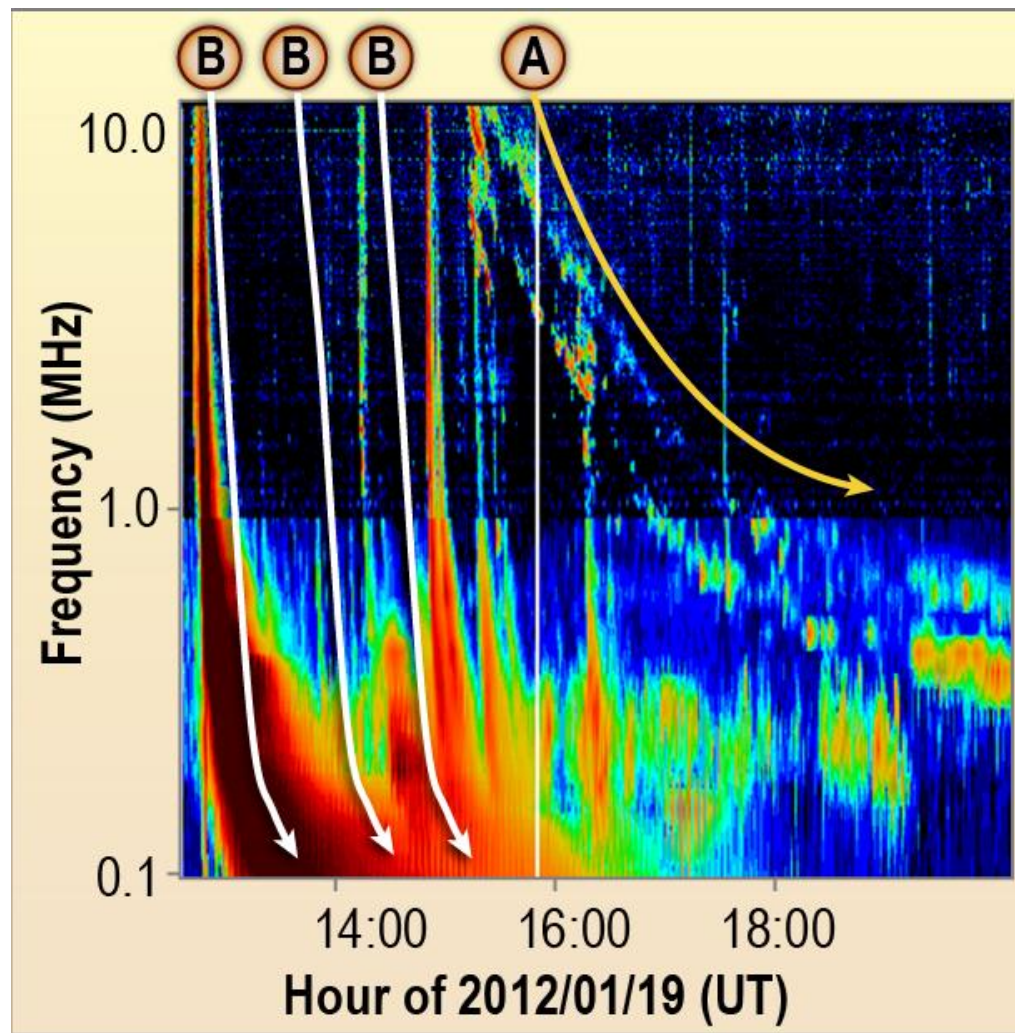


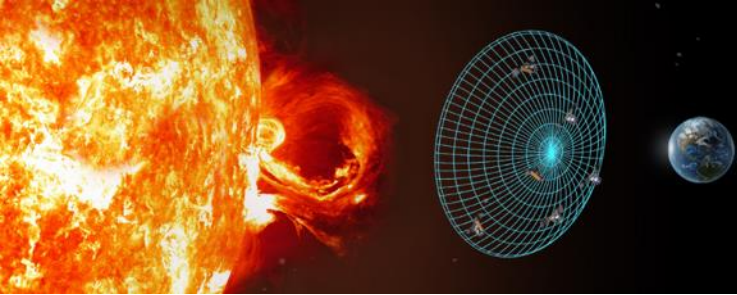


Solar Radio Bursts: Type II and III



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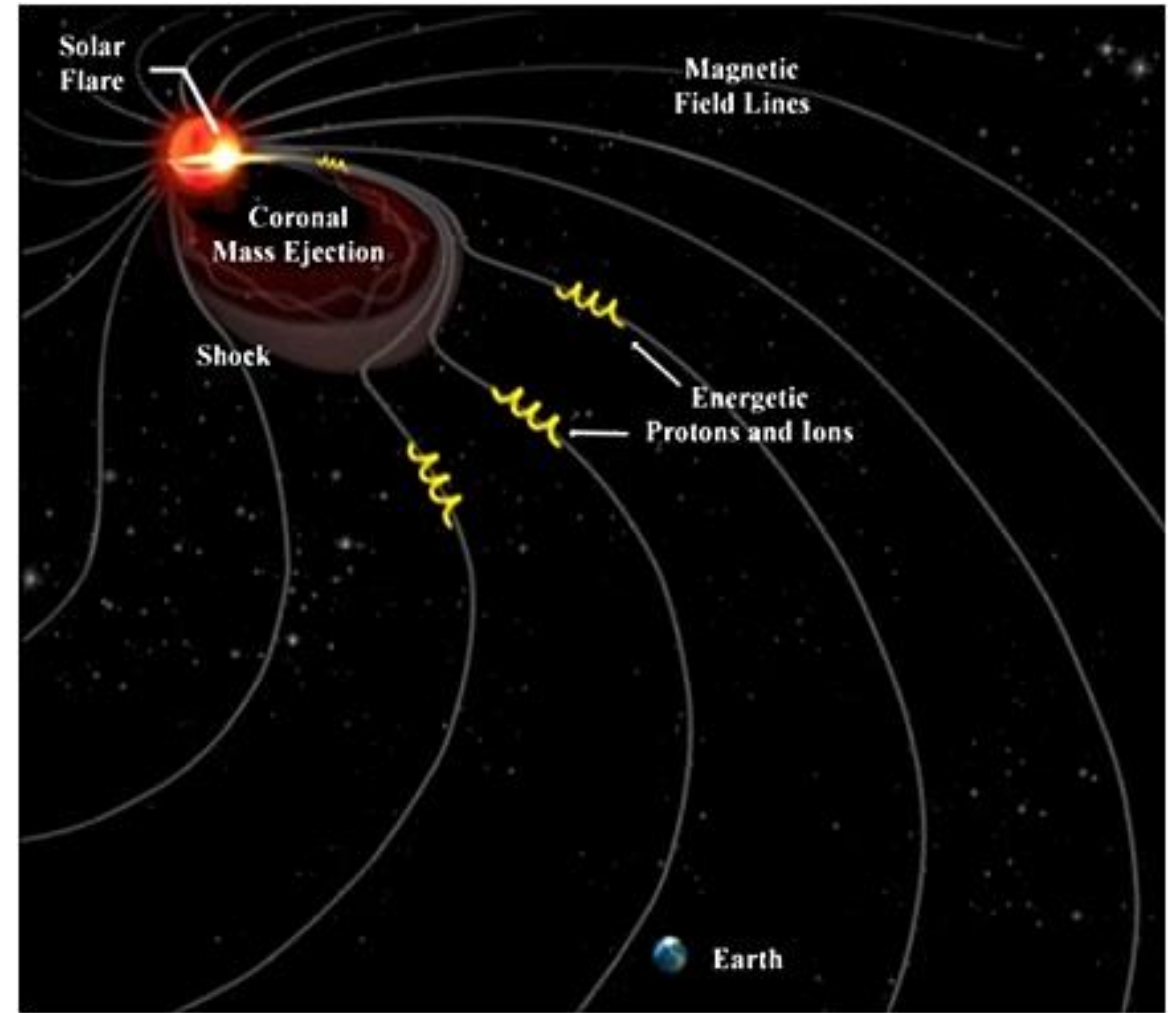
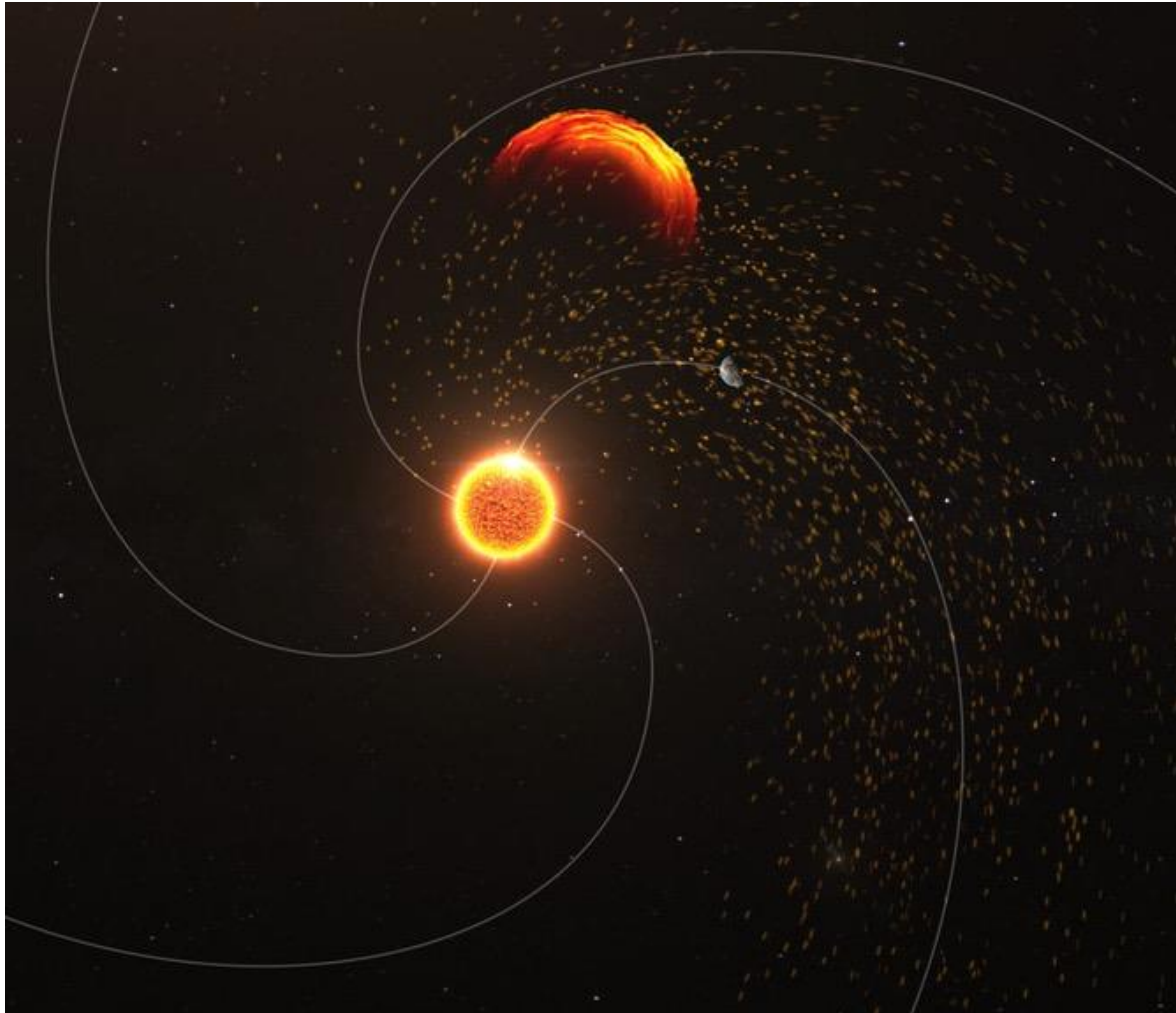


Solar Energetic Particles

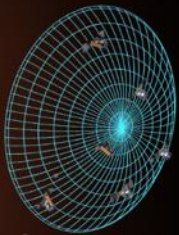


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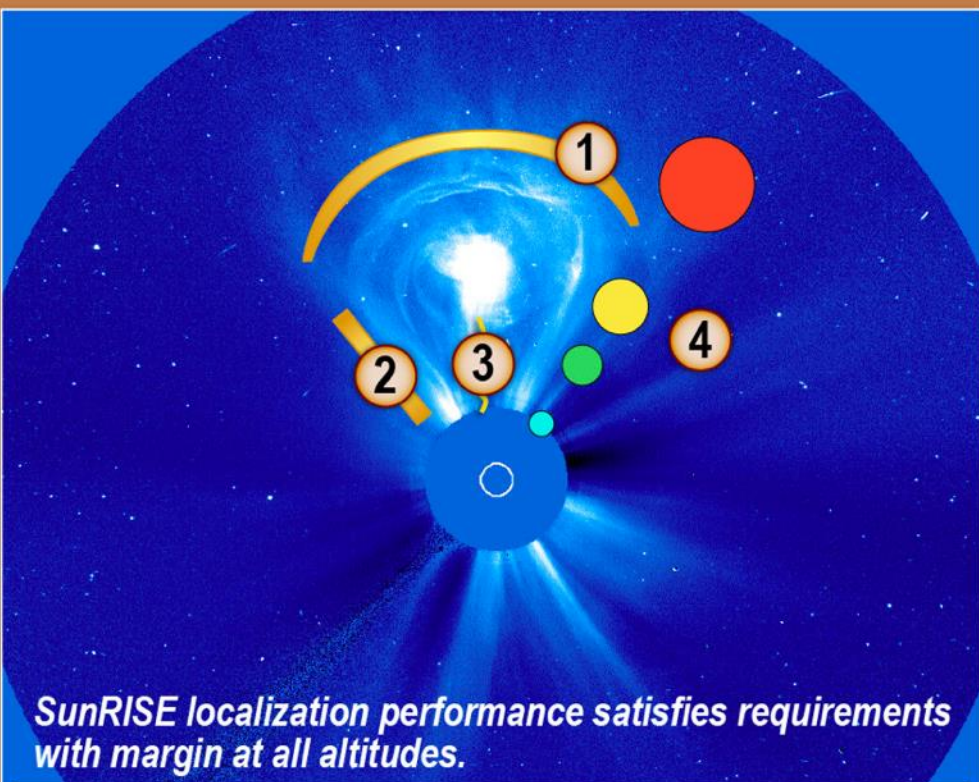
Source: <https://swatnet.eu/solar-energetic-particles/>



SunRISE Objectives



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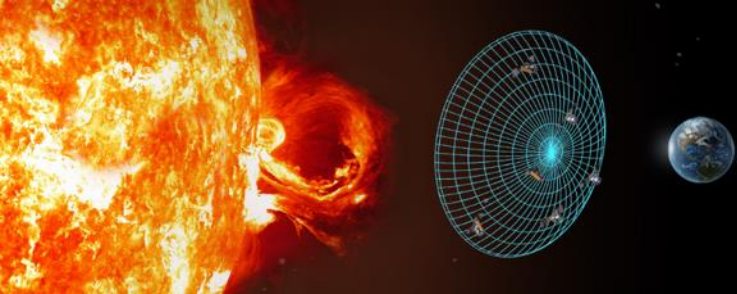


Localization Requirement	
4 R _S (ν = 1.0 MHz)	12 R _S (ν = 0.4 MHz)
8 R _S (ν = 0.8 MHz)	20 R _S (ν = 0.26 MHz)

Models RISE_062e

- 1 Parallel Shock Ahead of CME**
- 2 Perpendicular Shock from Flank Expansion**
- 3 Reconnection Behind CME**
- 4 Enhanced Turbulence from Compression**

Possible Shock Particles
 Magnetic Field

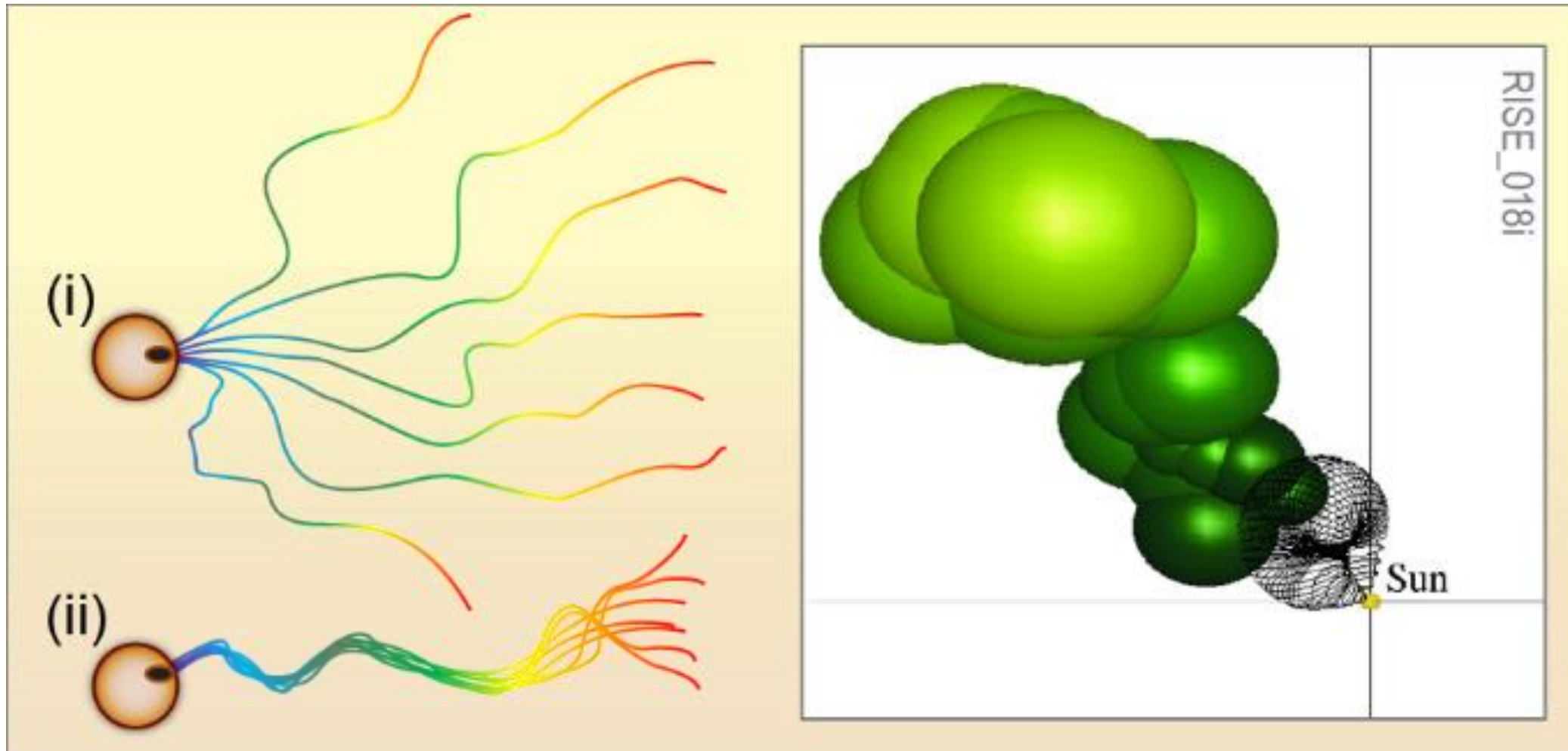


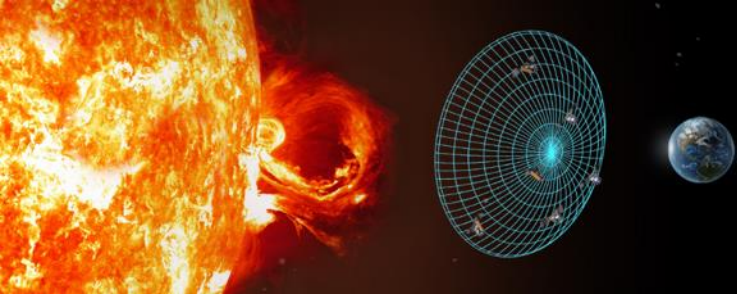
SunRISE Objectives



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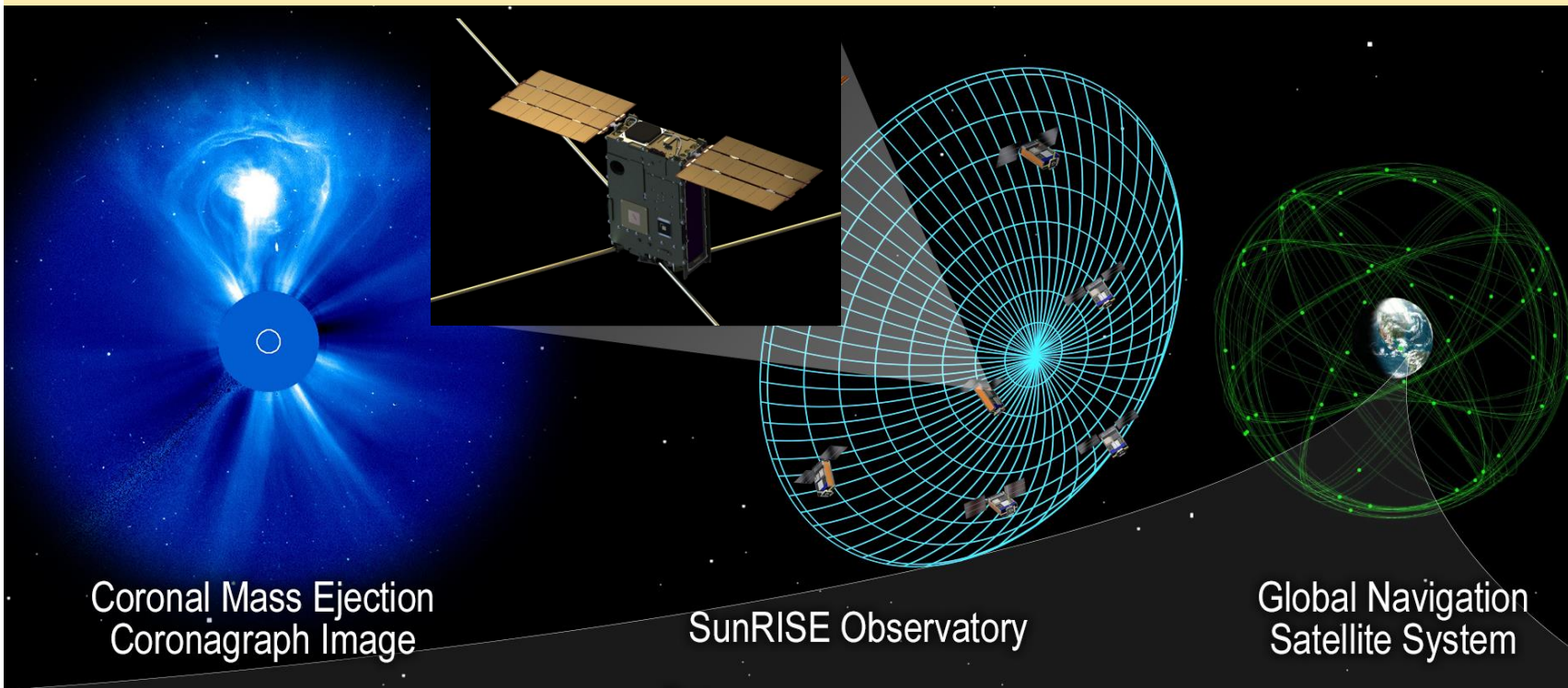


SunRISE Concept



SunRISE is the first spaceborne imaging radio interferometer

Uses existing technology to form a synthetic aperture for imaging Solar bursts at radio frequencies of 100 kHz to 23 MHz.



Coronal Mass Ejection
Coronagraph Image

SunRISE Observatory

Global Navigation
Satellite System

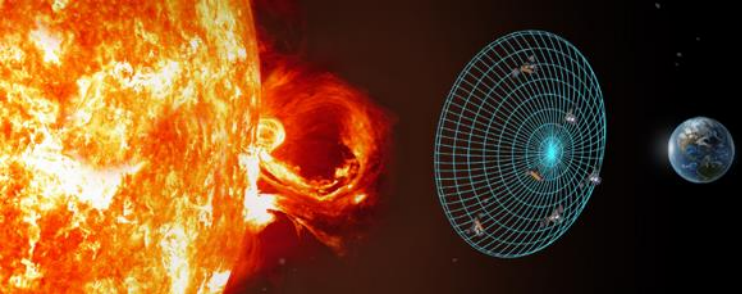
SunRISE images solar activity at decametric-hectometric (DH) radio frequencies.

Each space vehicle carries two orthogonal 5.5-m dipoles to measure the DH signals.

Interferometry at these frequencies needs position and time knowledge to 1m and 1ns.

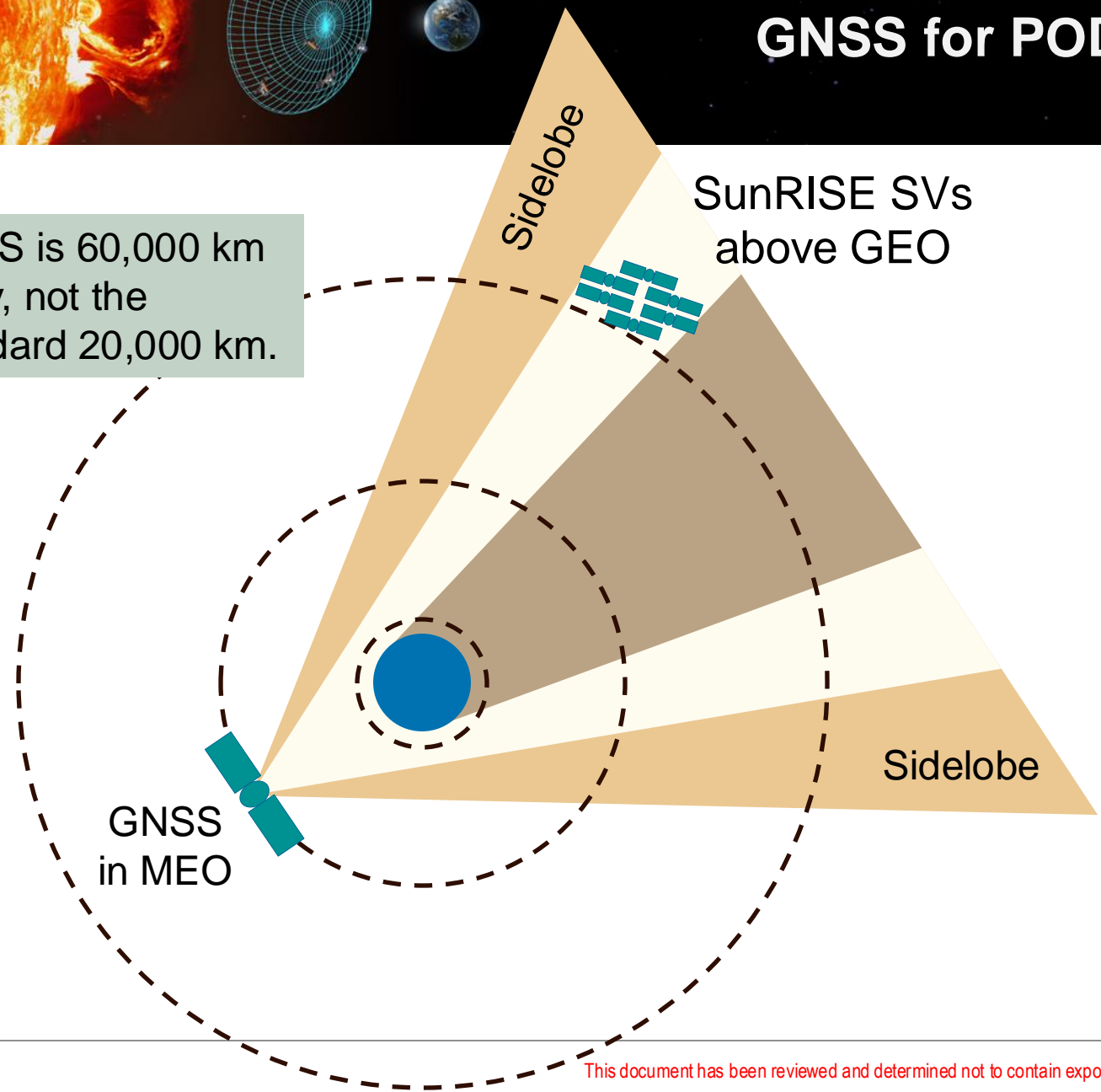
Frequencies below 23 MHz can be distorted by the ionosphere, so SunRISE flies just above GEO.

Orbital period is 3-4 months.



GNSS for POD

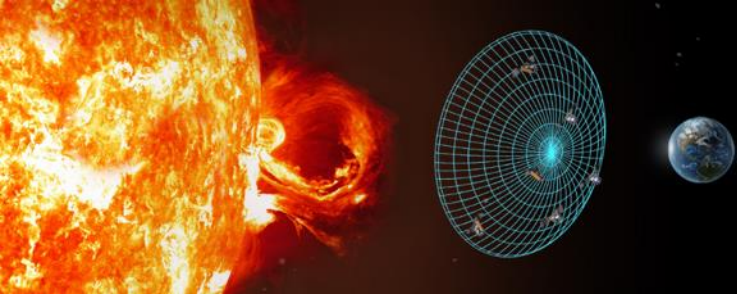
GNSS is 60,000 km away, not the standard 20,000 km.



The GNSS signal chain achieves on-board knowledge of time (and position) to a microsecond. This allows all 6 space vehicles to take simultaneous DH measurements.

Post processing achieves higher accuracy Precision Orbital Determination (POD) solutions, which are used to determine the relative propagation delays between the space vehicle pairs in the interferometer.

POD is also used for navigation planning and Forward Orbit Prediction.



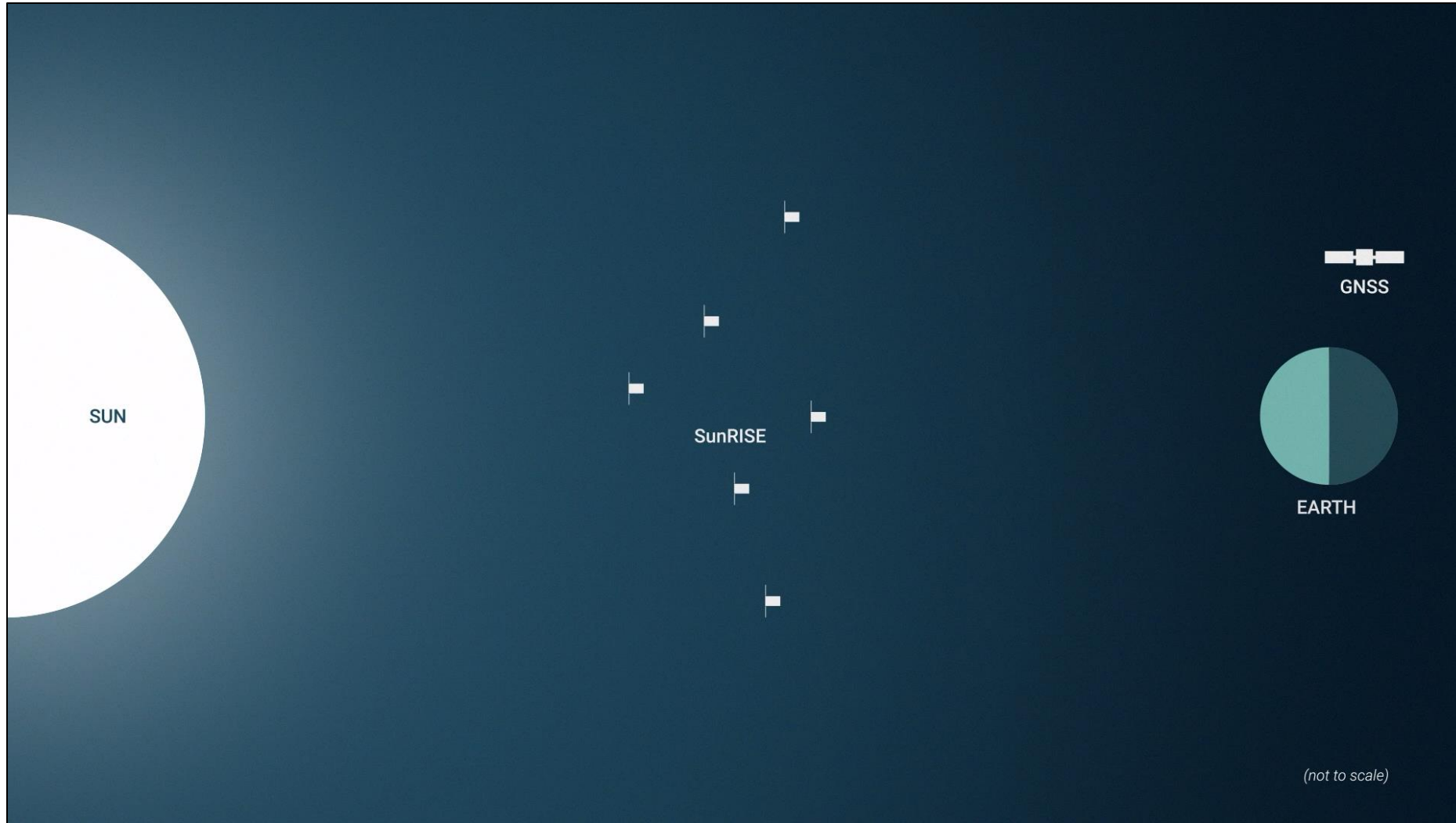
Observatory

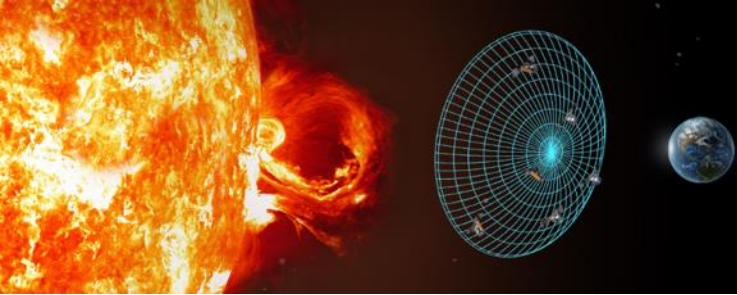


GNSS provides
~1 m position
accuracy on each
spacecraft.

Supersynchronous
orbit 37,000 km
altitude.

3-20 km spacecraft
separations

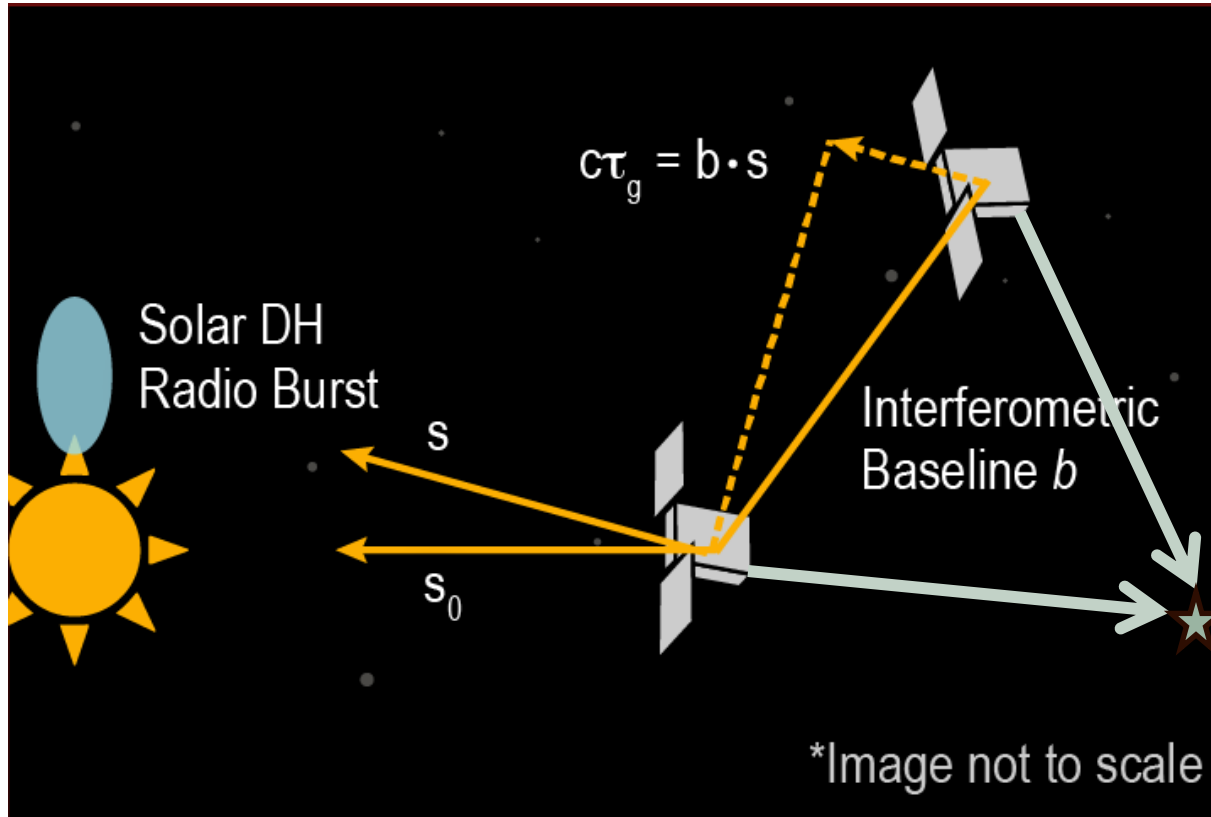




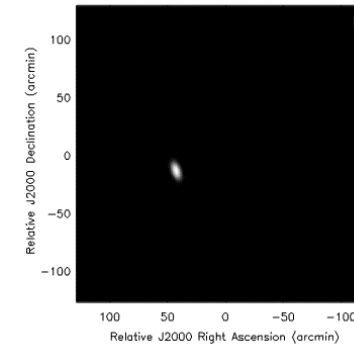
Interferometer for Imaging



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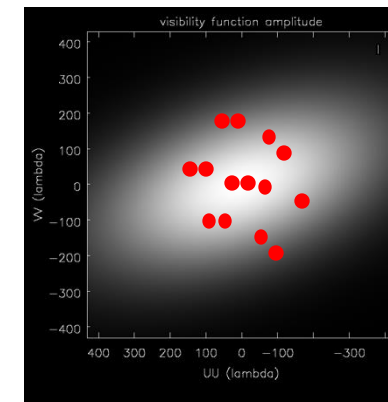
Basic Interferometric Element



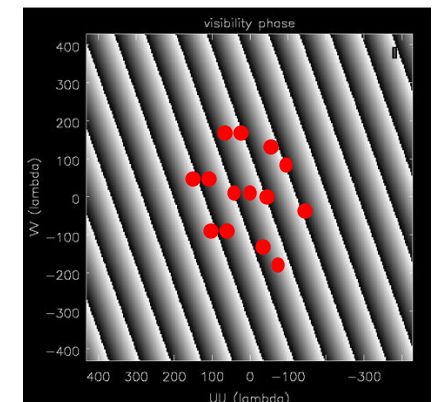
SunRISE combines visibilities from all unique baselines (15 for N=6)

Requires knowing position and time accurately (~1m, ~1ns)

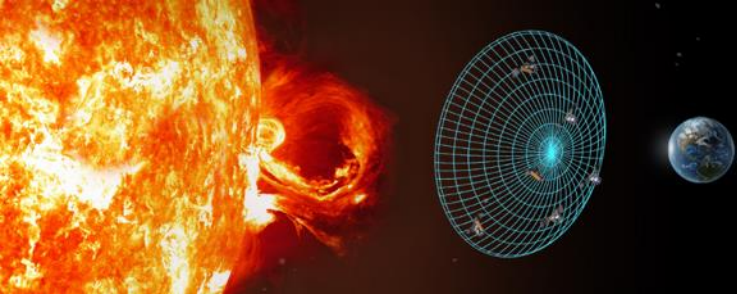
Fourier transform



amplitude

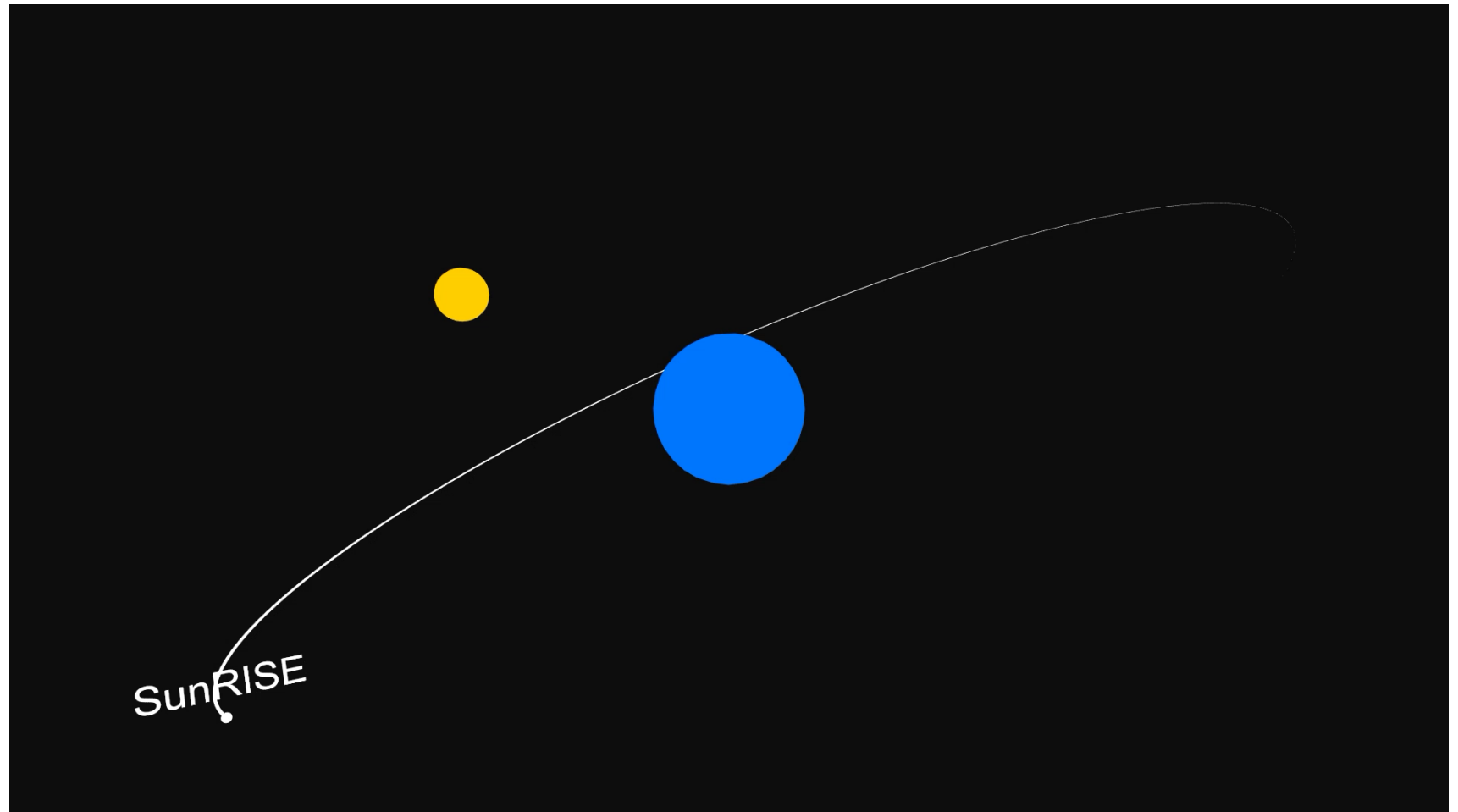


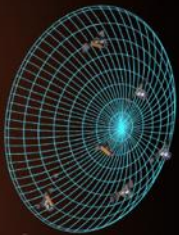
phase



Orbits and Baseline Coverage

- 6.5 km radius disk in the projected plane of the Sun
- SunRISE orbits are designed so that 2 orthogonal projected baseline lengths are >6.5 km.
- This enables the resolution needed.
- SunRISE is a digitally steerable radio telescope allowing for comparable resolution anywhere in the sky.

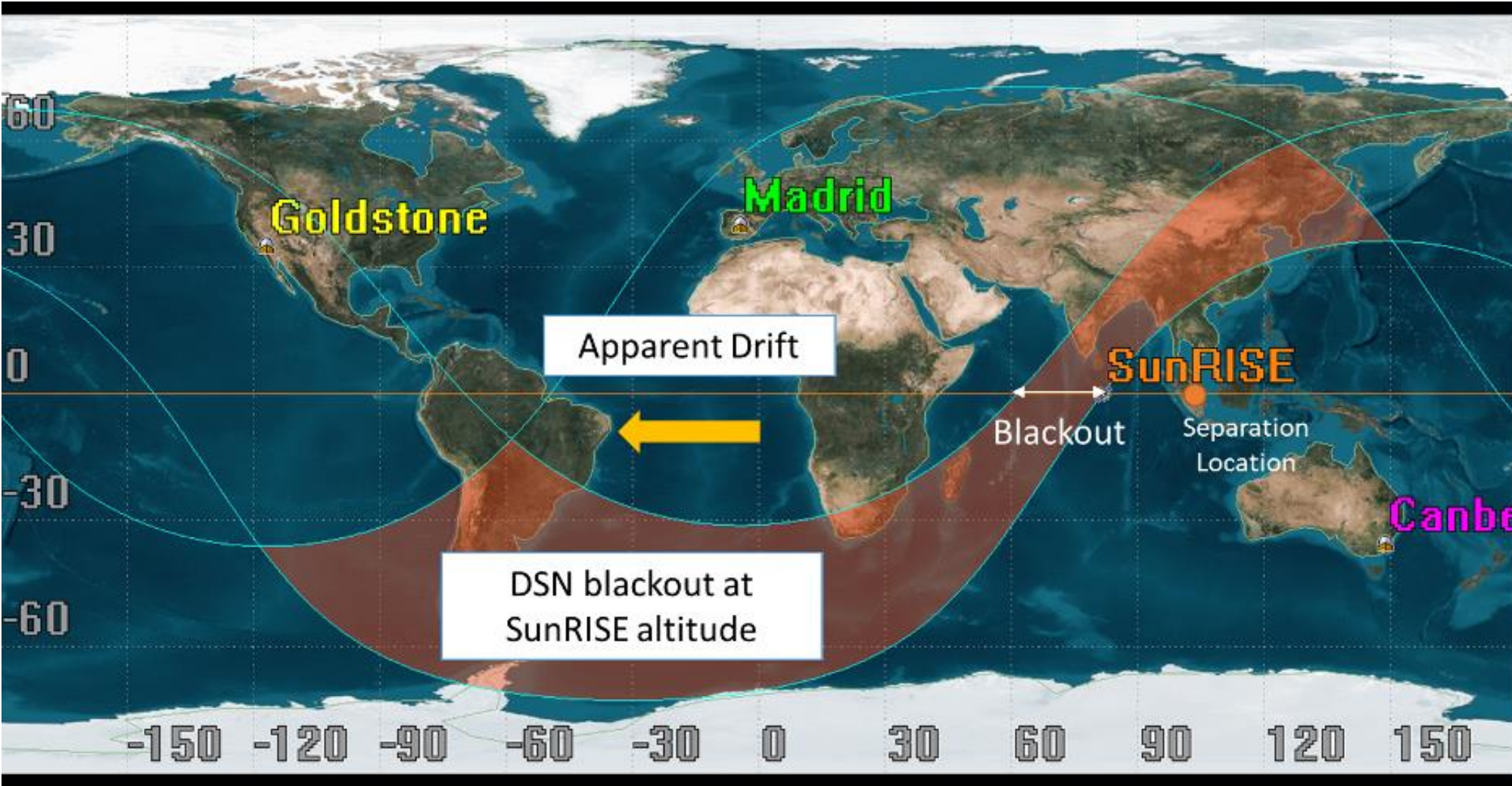


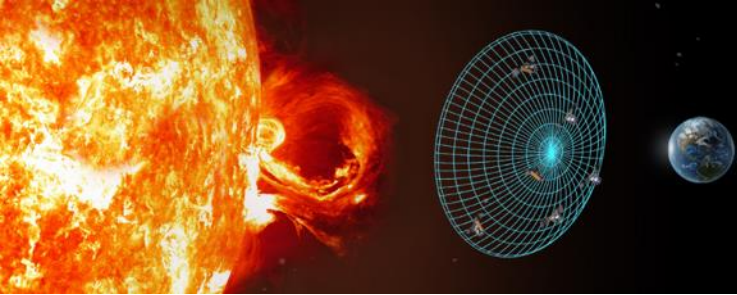


Ground Communications Coverage



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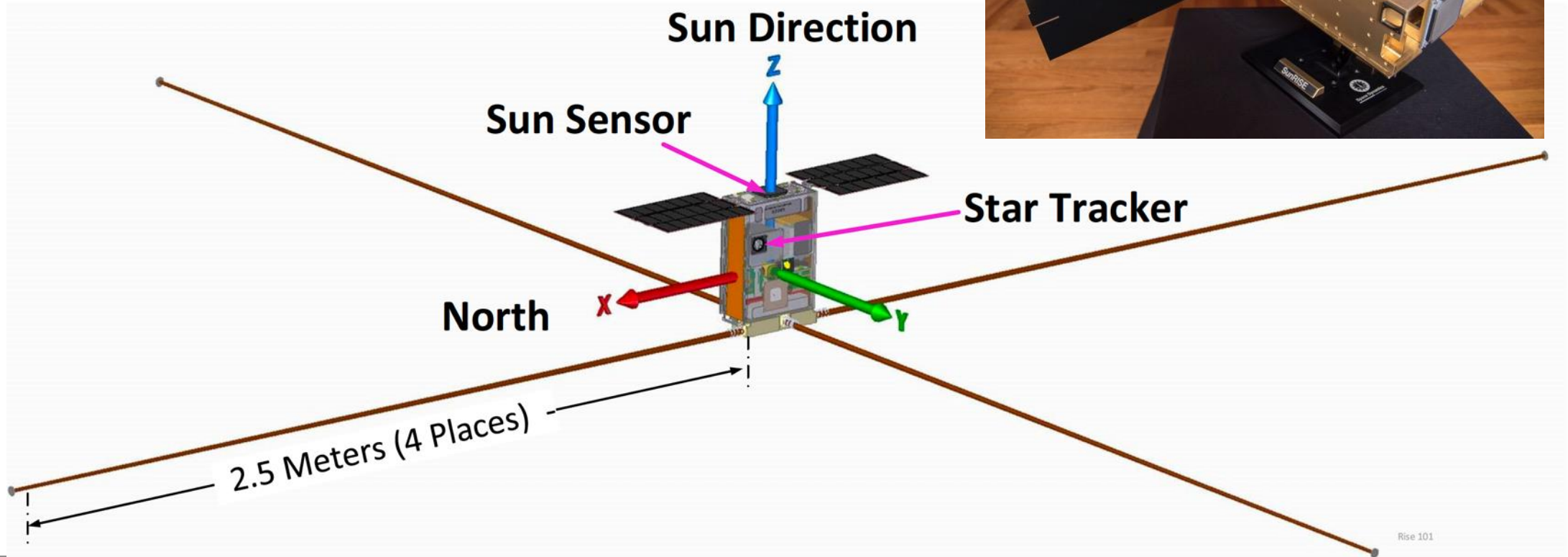


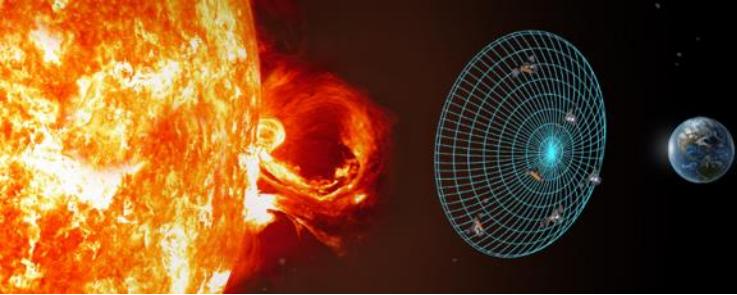


SunRISE Spacecraft



- Electrically short antenna for 0.1-25 MHz deca-hectometric (DH) receiver.
- Dual-polarized receiver.
- Integrated DH and GNSS receiver fits in 6U CubeSat form factor.





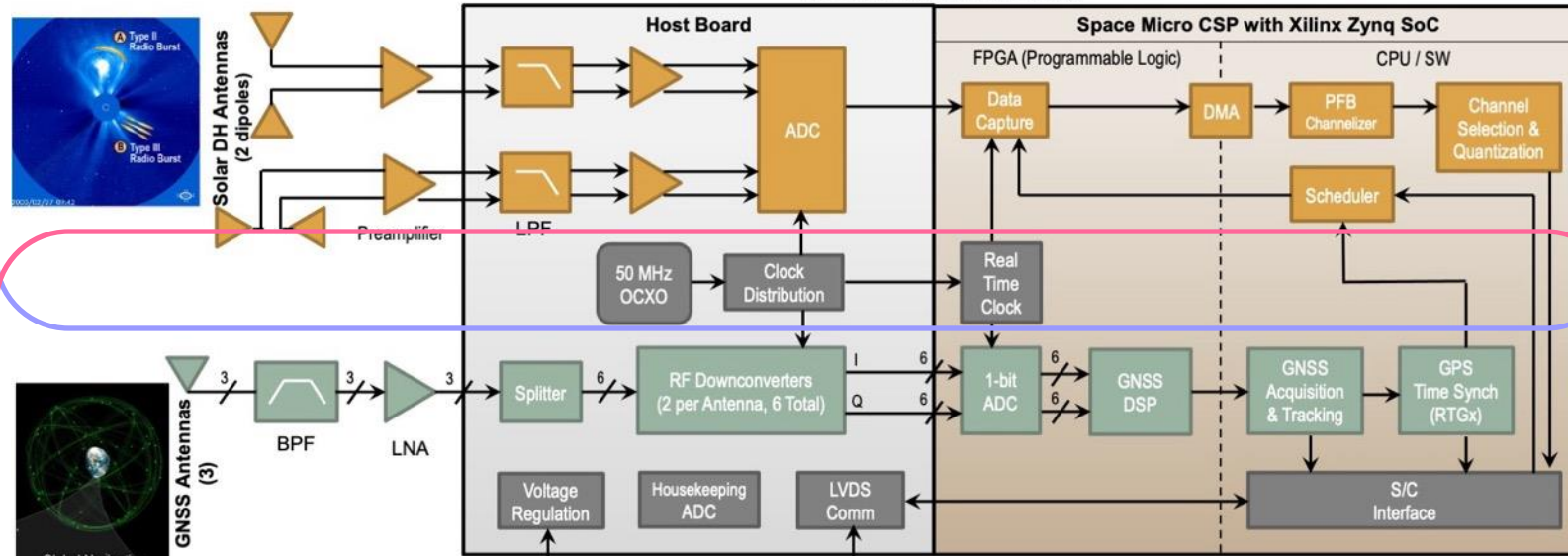
Science Payload



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Measures the burst

DH-GNSS Receiver Assembly

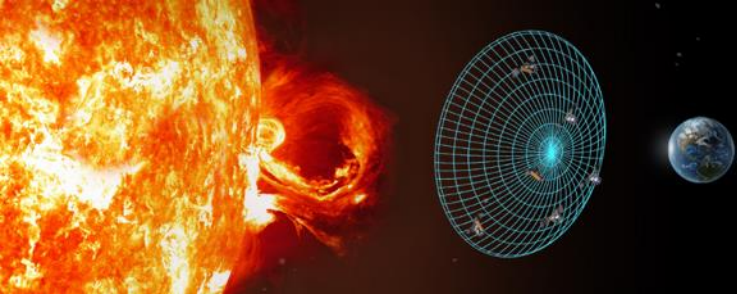


Tells where and when we made the measurement

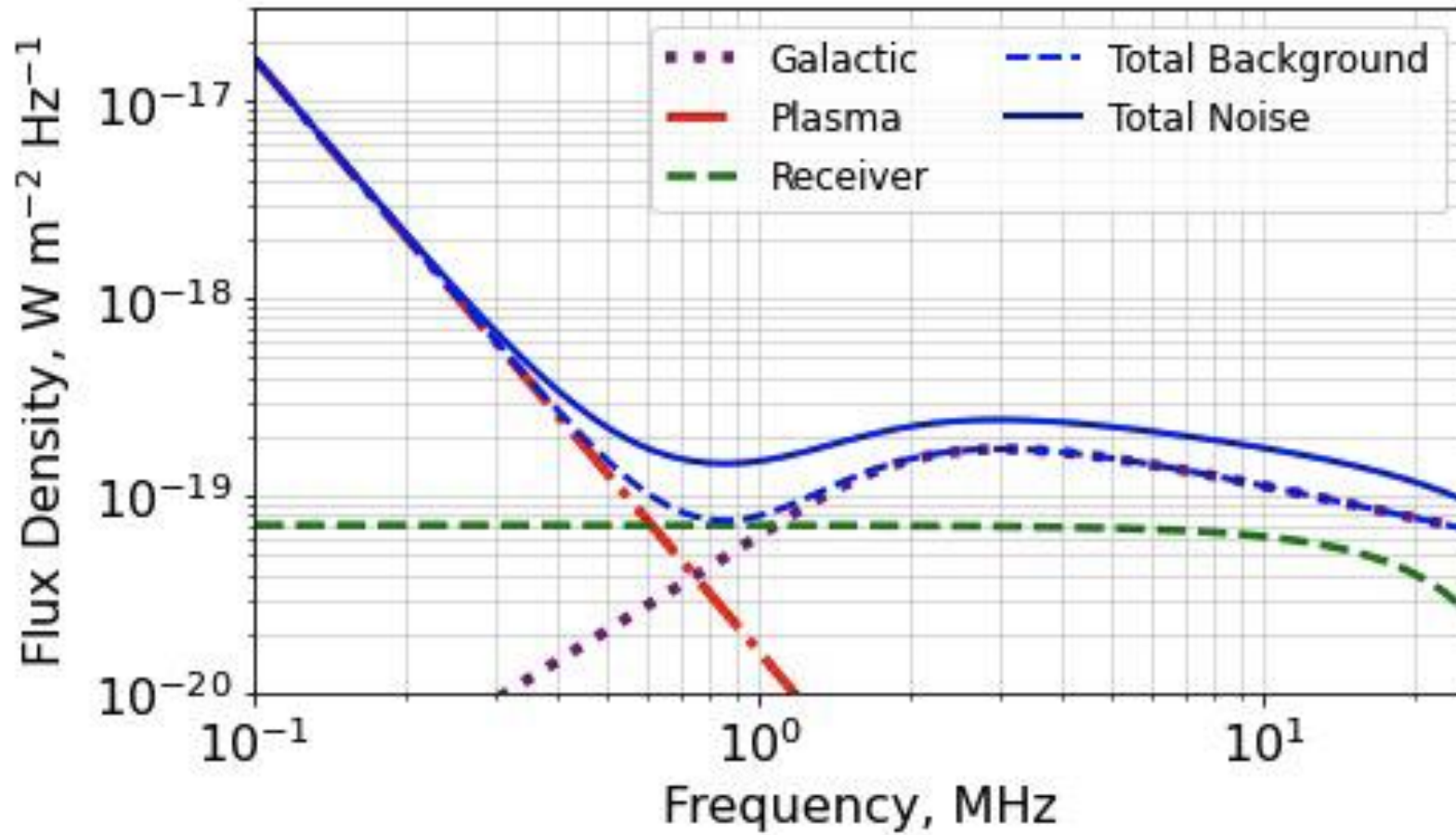
Common clock for GNSS and DH measurements connects the measurement to a time and place

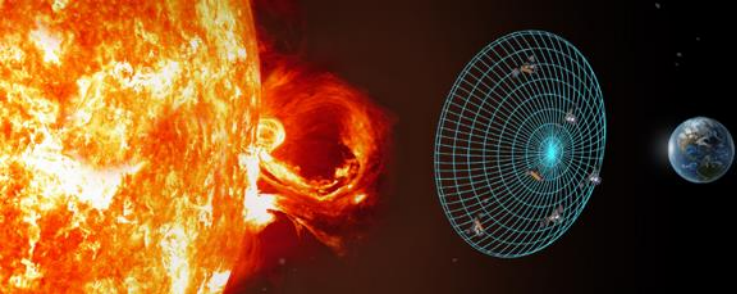


All six Payloads during Integration and Test (I&T)

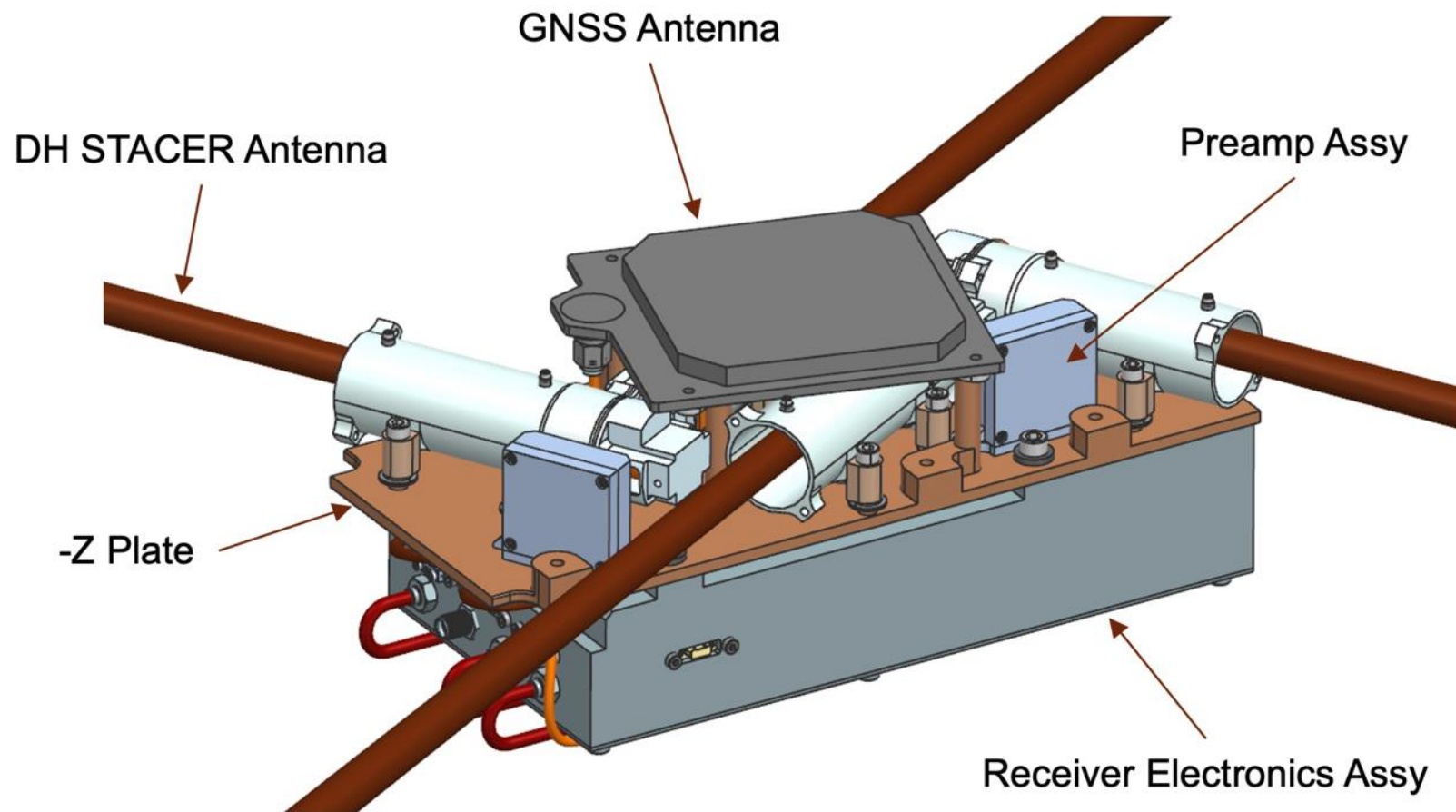


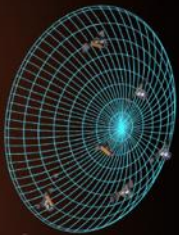
Payload Sensitivity





Payload





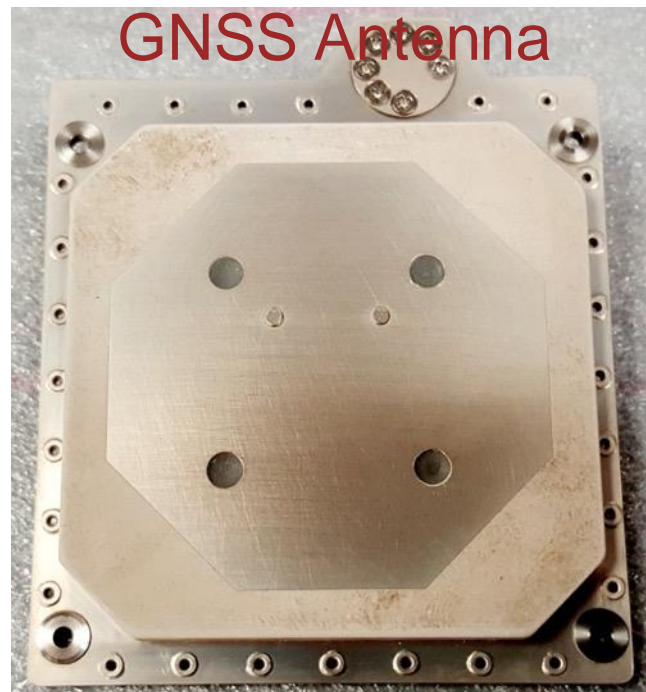
Payload Hardware



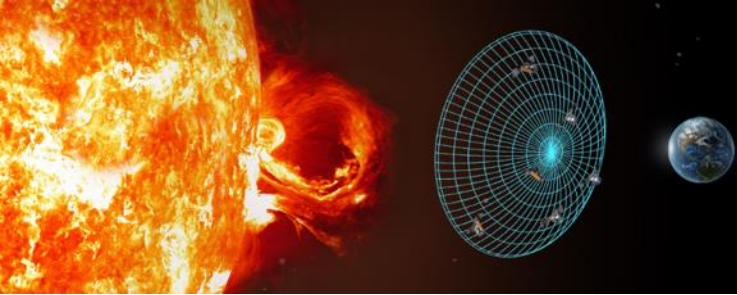
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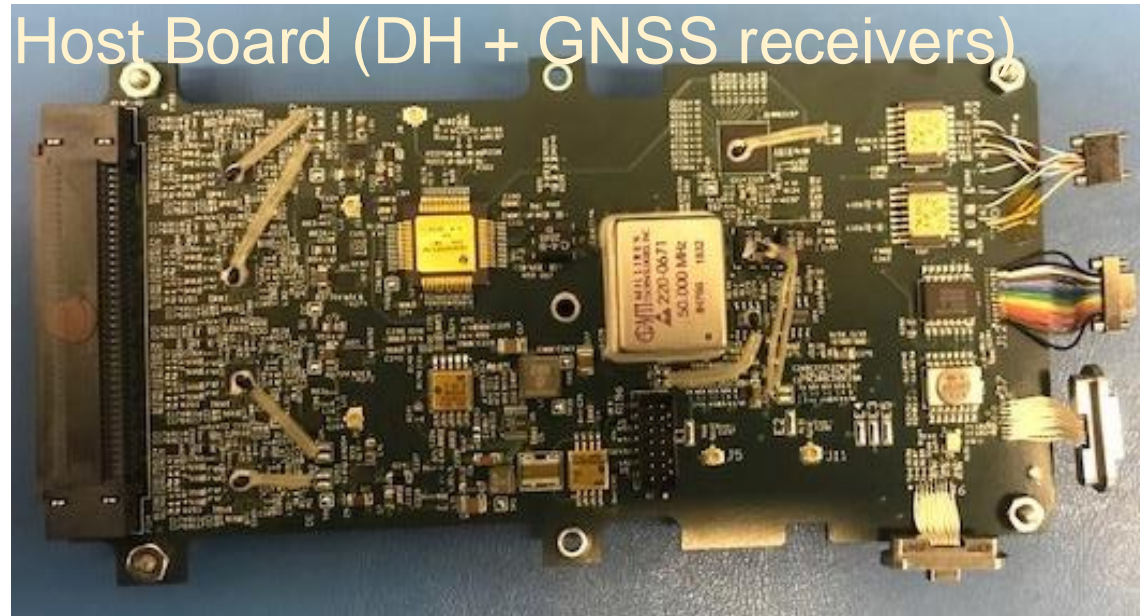
Payload

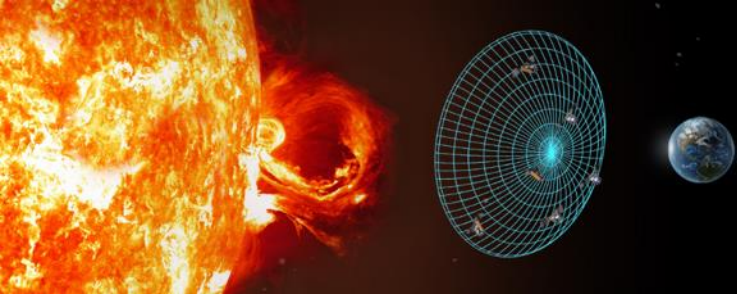


GNSS Antenna

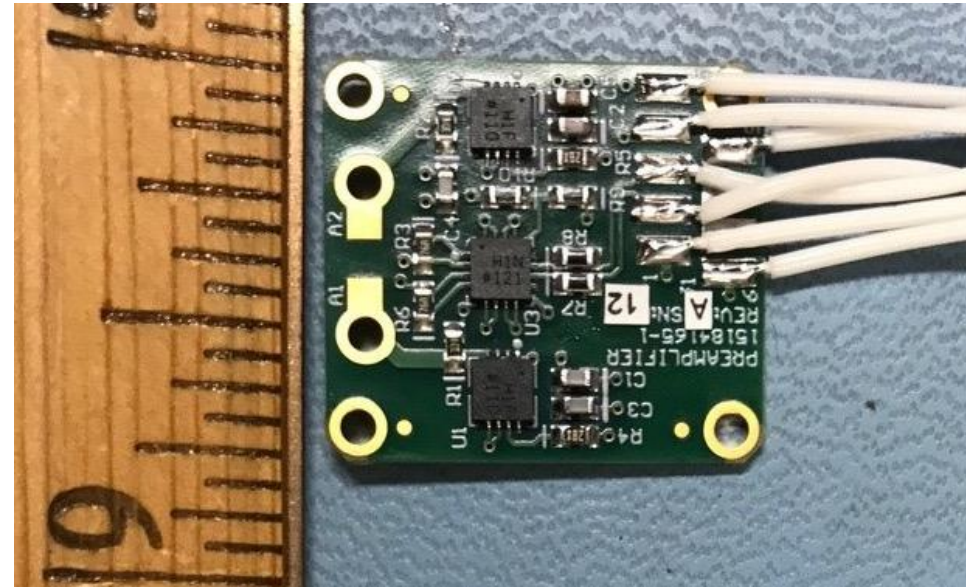
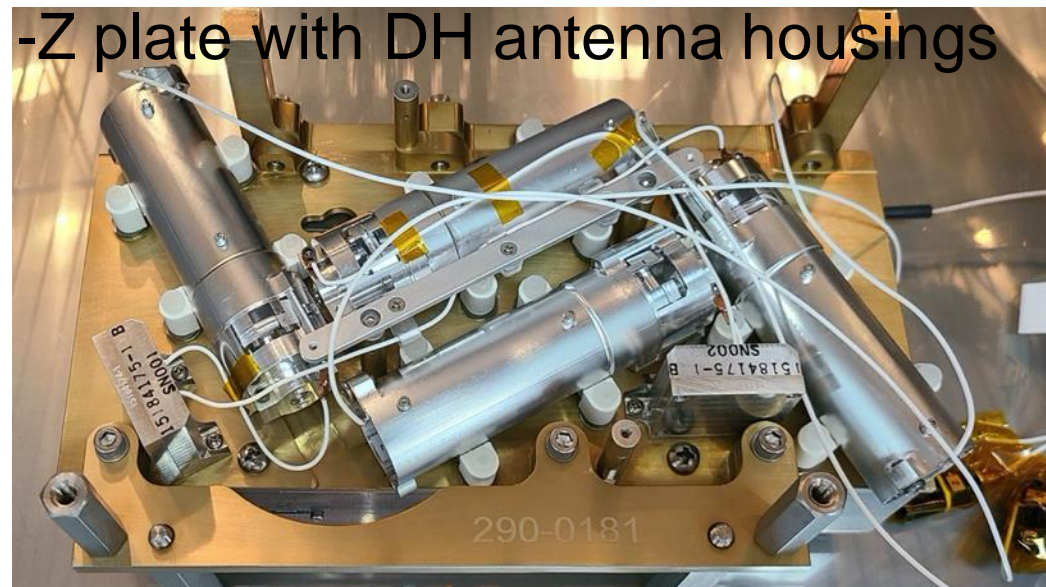


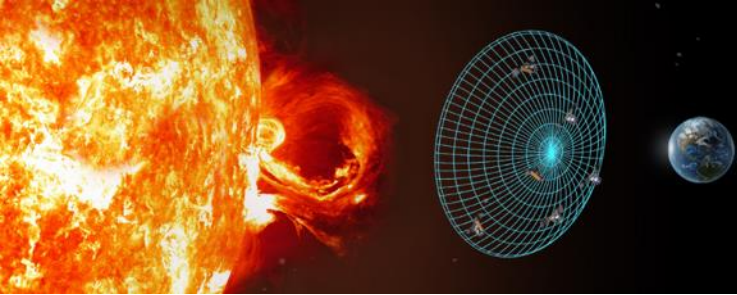
Payload Hardware





Payload Hardware



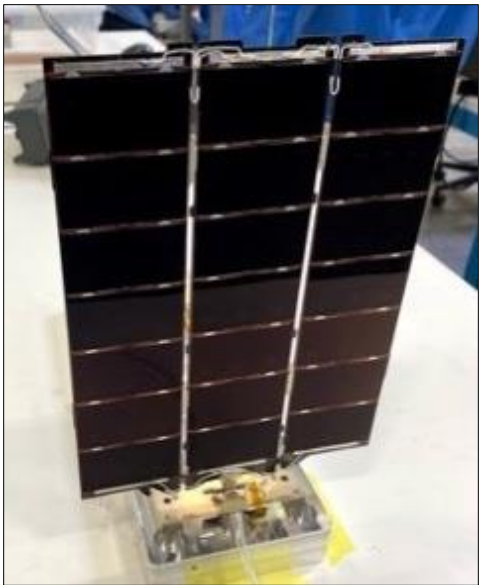


Flight System



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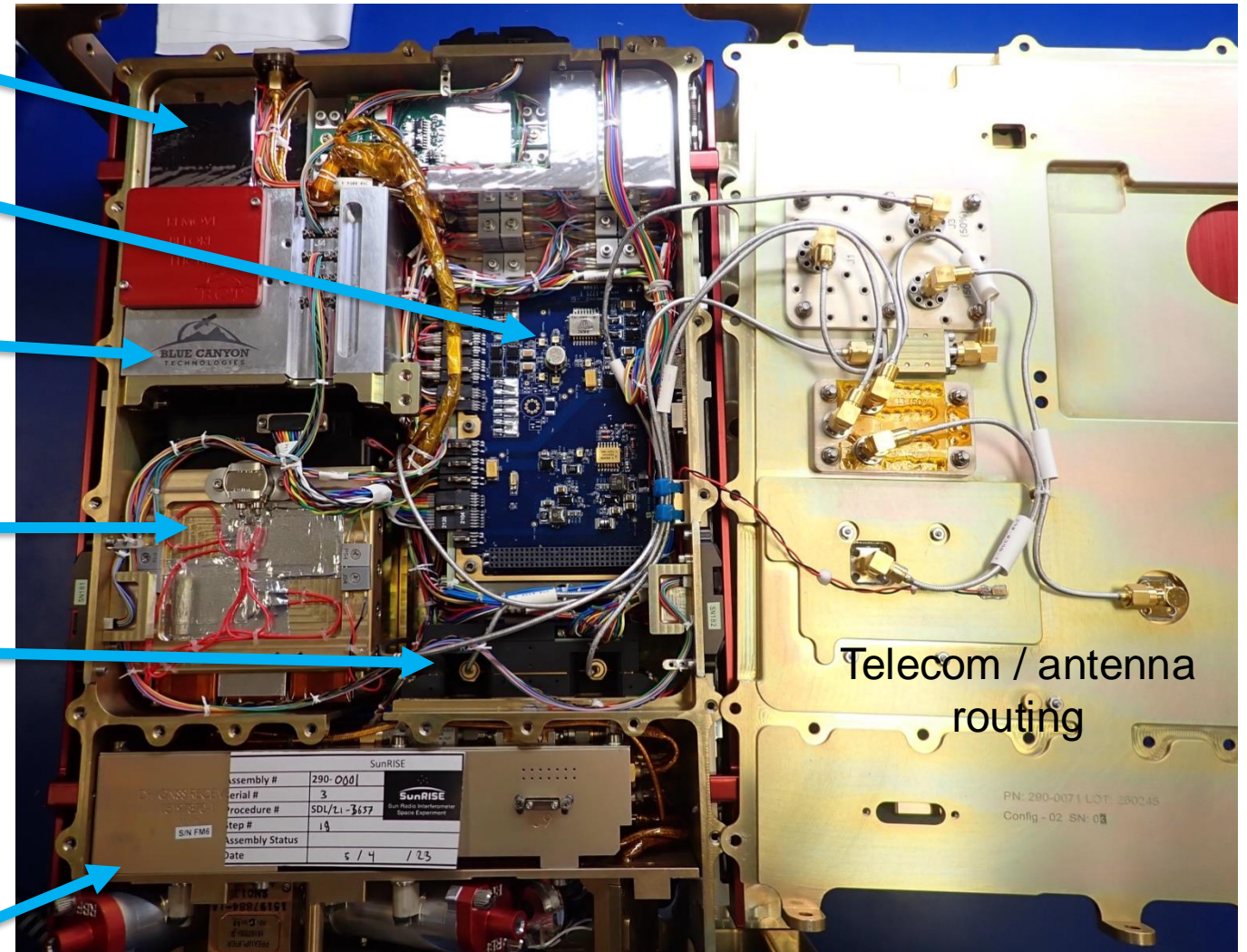
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Solar Panel (2x)

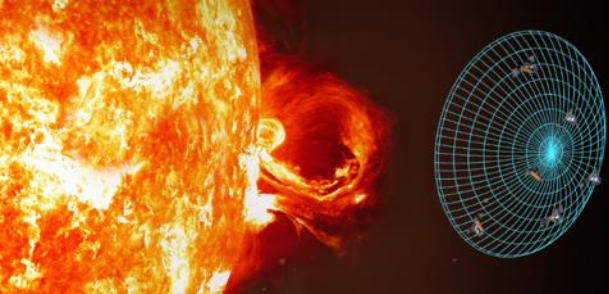
About 12 kg in 6 liters is $\sim 2\text{g/cm}^3$
 Compare Aluminum 2.7 g/cm^3
 Like most cubesats, we run out of volume before we run out of mass
 (Dispenser is max 14 kg allowable)

- Propulsion
- Avionics
- Attitude Determination & Control
- Battery Pack
- Telecom Radio
- Solar DH-GNSS Payload



Telecom / antenna routing

SunRISE	
Assembly #	290-0001
Serial #	3
Procedure #	SDL/L1-3657
Step #	15
Assembly Status	
Date	5 / 4 / 23



NASA's 6-Pack of Mini-Satellites Ready for Their Moment in the Sun



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4 MIN READ

NASA's 6-Pack of Mini-Satellites Ready for Their Moment in the Sun

Jet Propulsion Laboratory

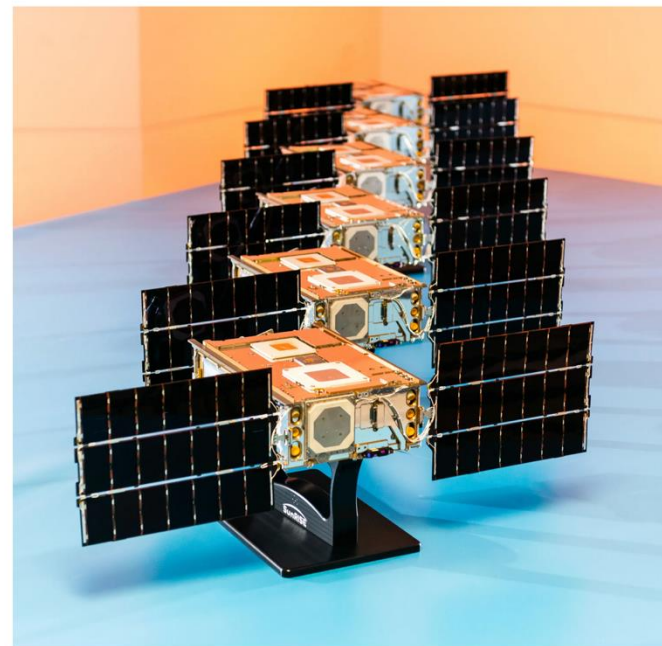
NOV 30, 2023

ARTICLE



CONTENTS

- Monitoring Solar Radio Bursts
- More About the Mission



The six satellites that make up NASA's SunRISE mission are each only about the size of a cereal box, flanked by small solar panels. This fleet of six SmallSats will work together to effectively create a much larger radio antenna in space. Space Dynamics Laboratory/Allison Bills

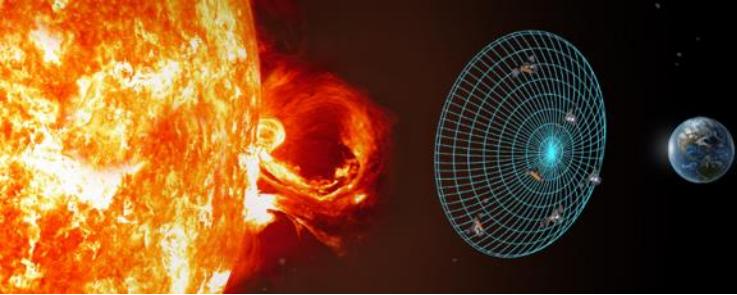
No, they do NOT fly this close together in space! 1km is a "close approach"



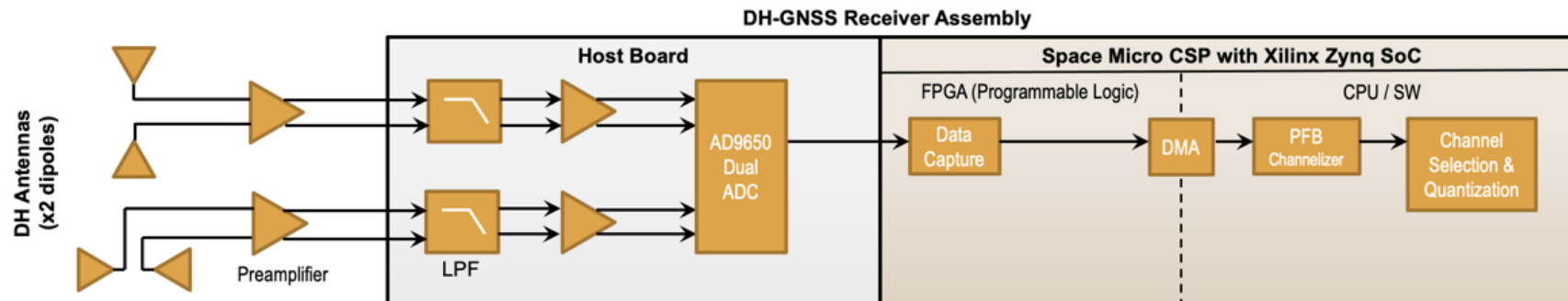
All the SVs snug in their storage

Credit: SDL/Allison Bills

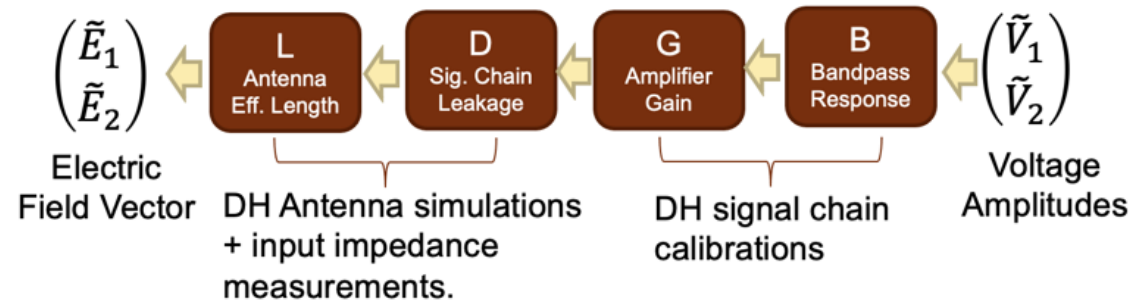
Launch No Earlier Than Sep 2024
More likely in mid-late 2025



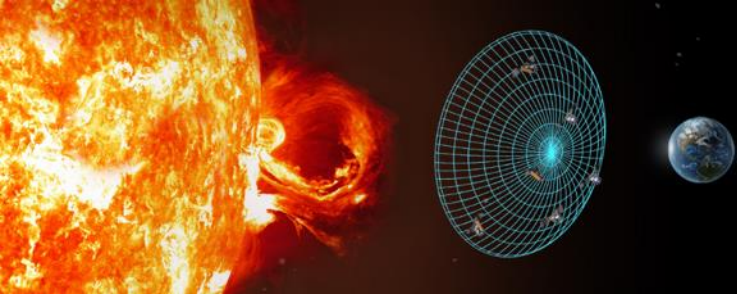
Measurements



DH Signal Chain Calibration Matrices
Backing out electric field from voltages



- DH signal chain acquires at a rate of 10 Hz with 6.1 kHz channels.
- End-to-end signal chain calibration for antenna effective length, signal chain leakage, gain, and bandpass.
- DH + GNSS data telemetered to ground for interferometric processing

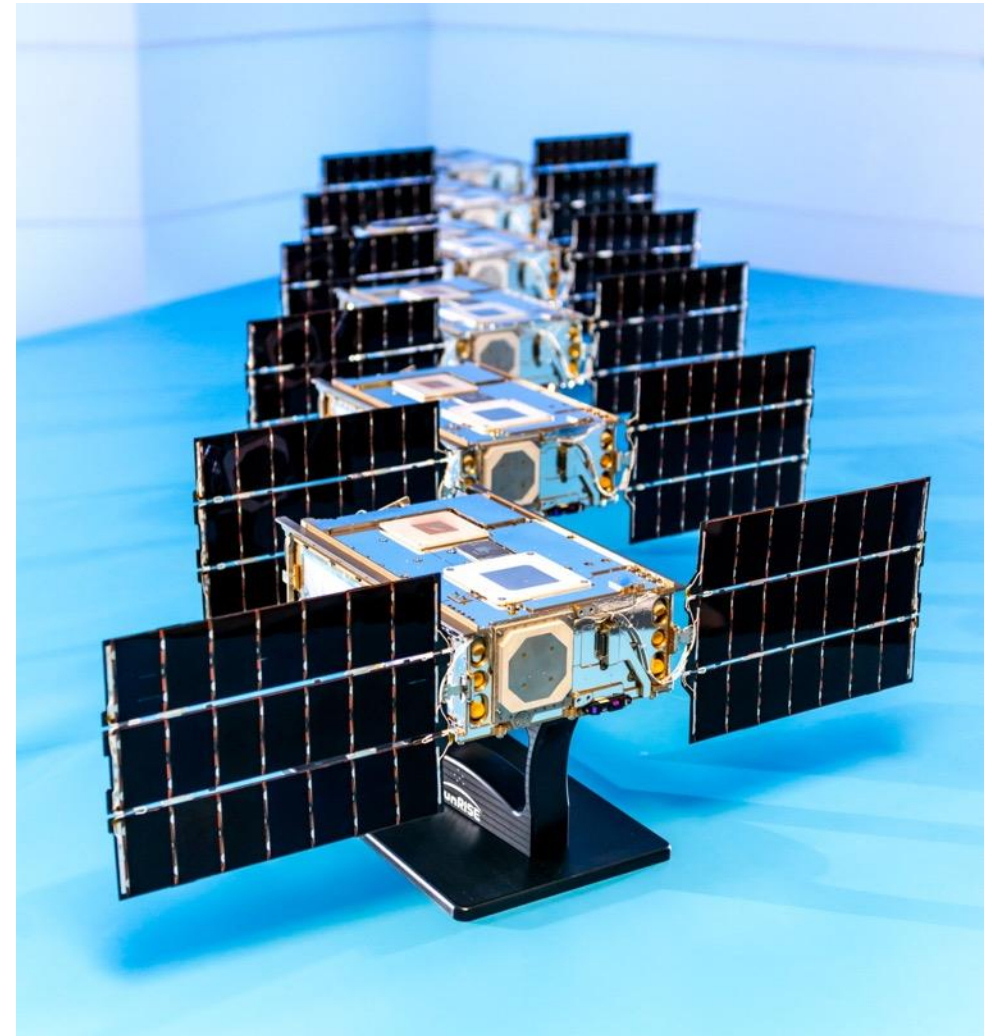
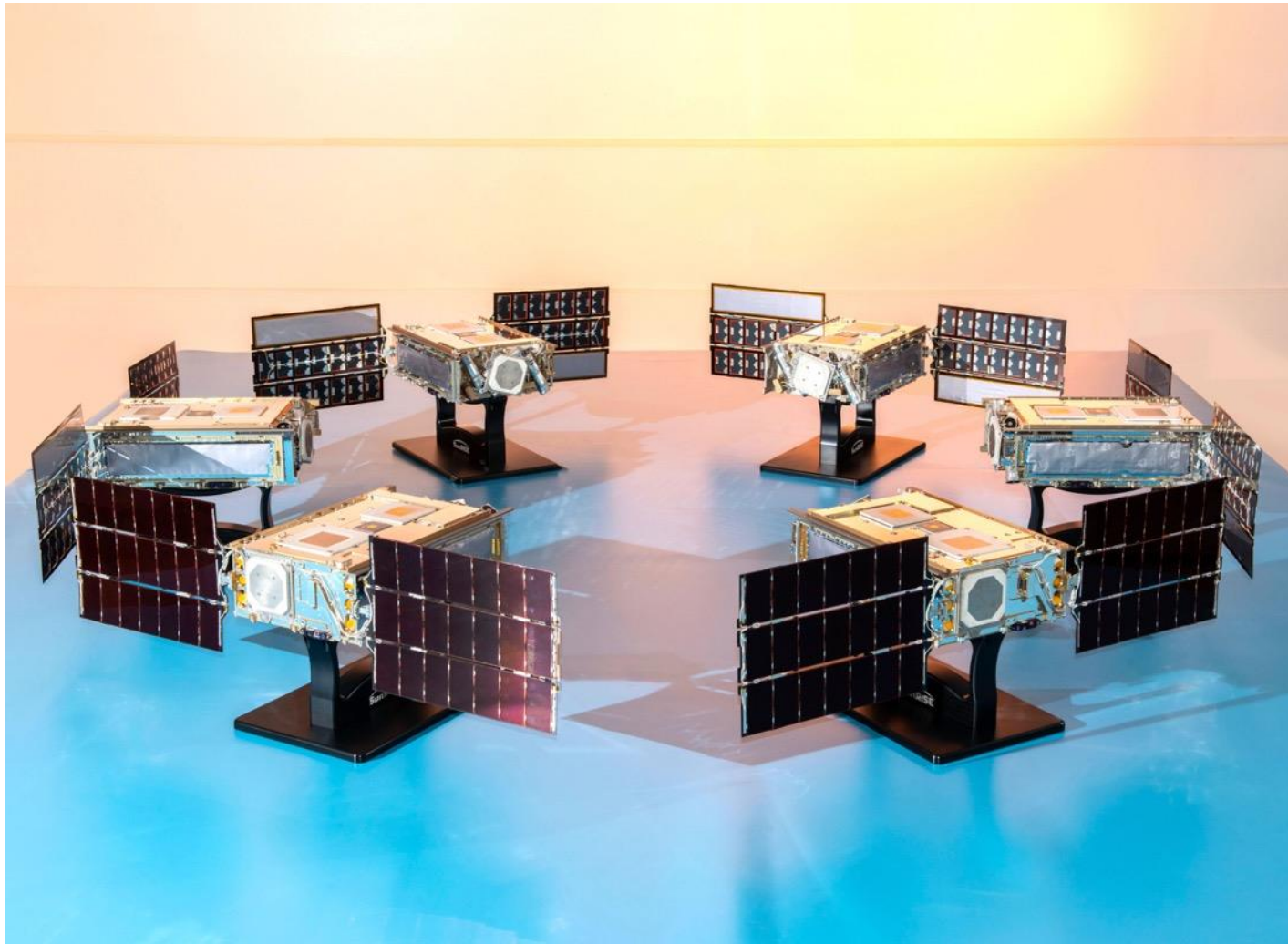


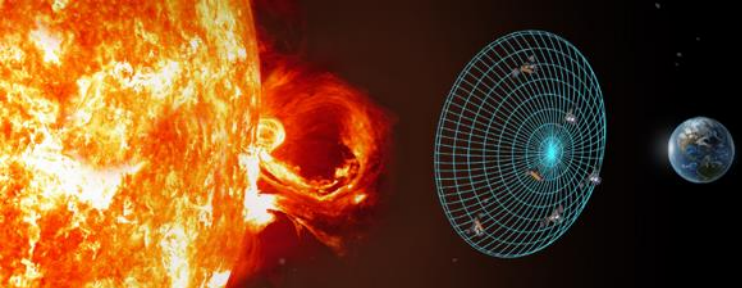
SunRISE Ready to Go



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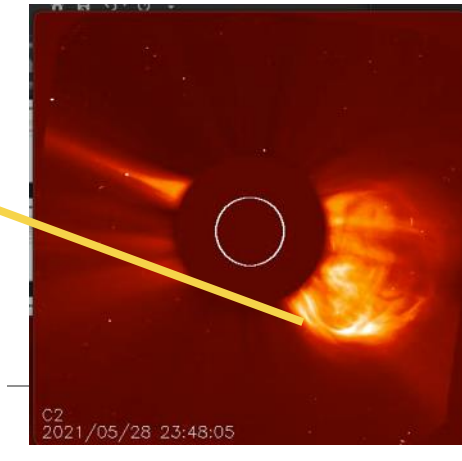
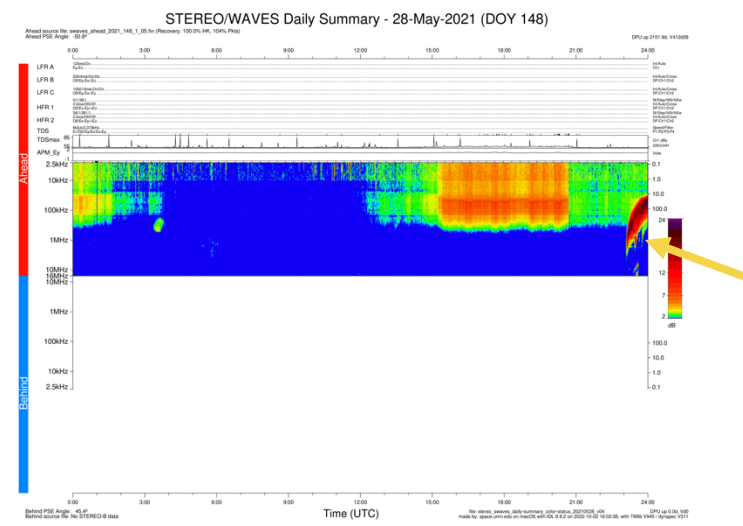
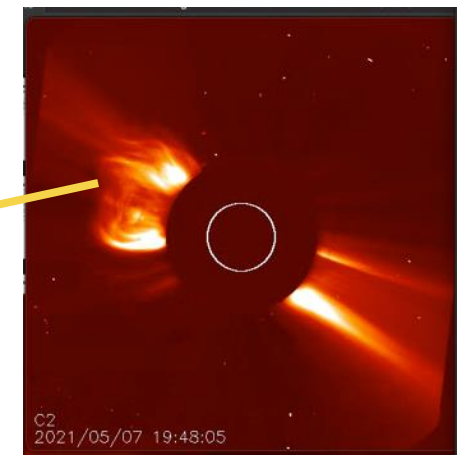
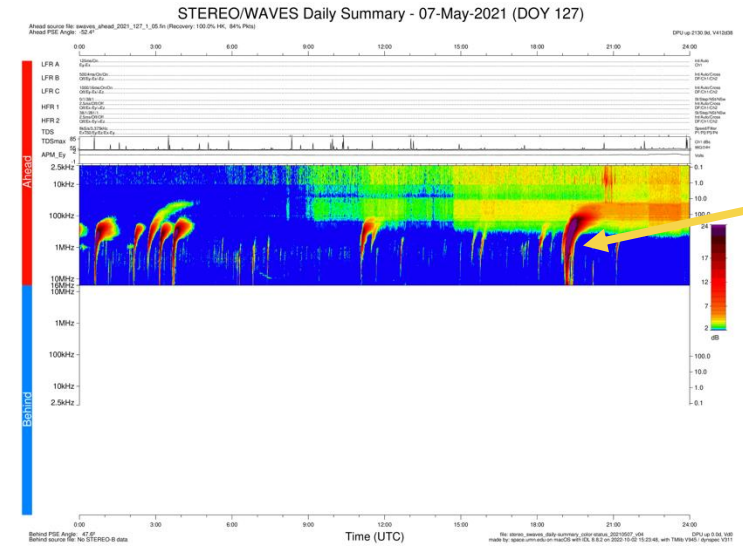
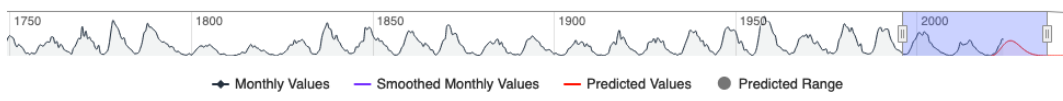
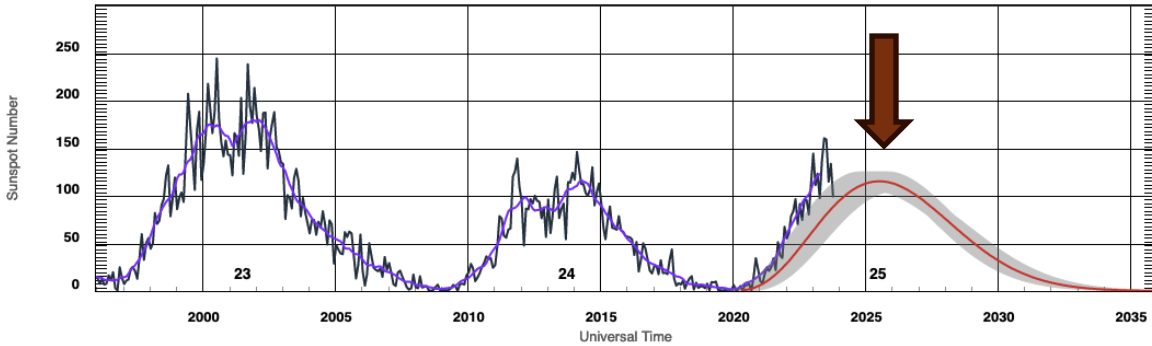
Solar Cycle 25 CME-Associated Type II Bursts

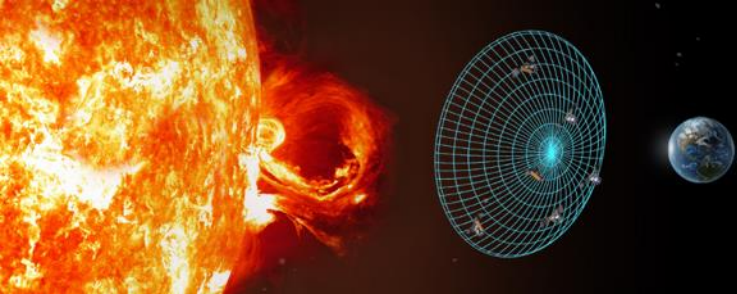


SUNRISE
SUN RADIO INTERFEROMETER SPACE EXPERIMENT

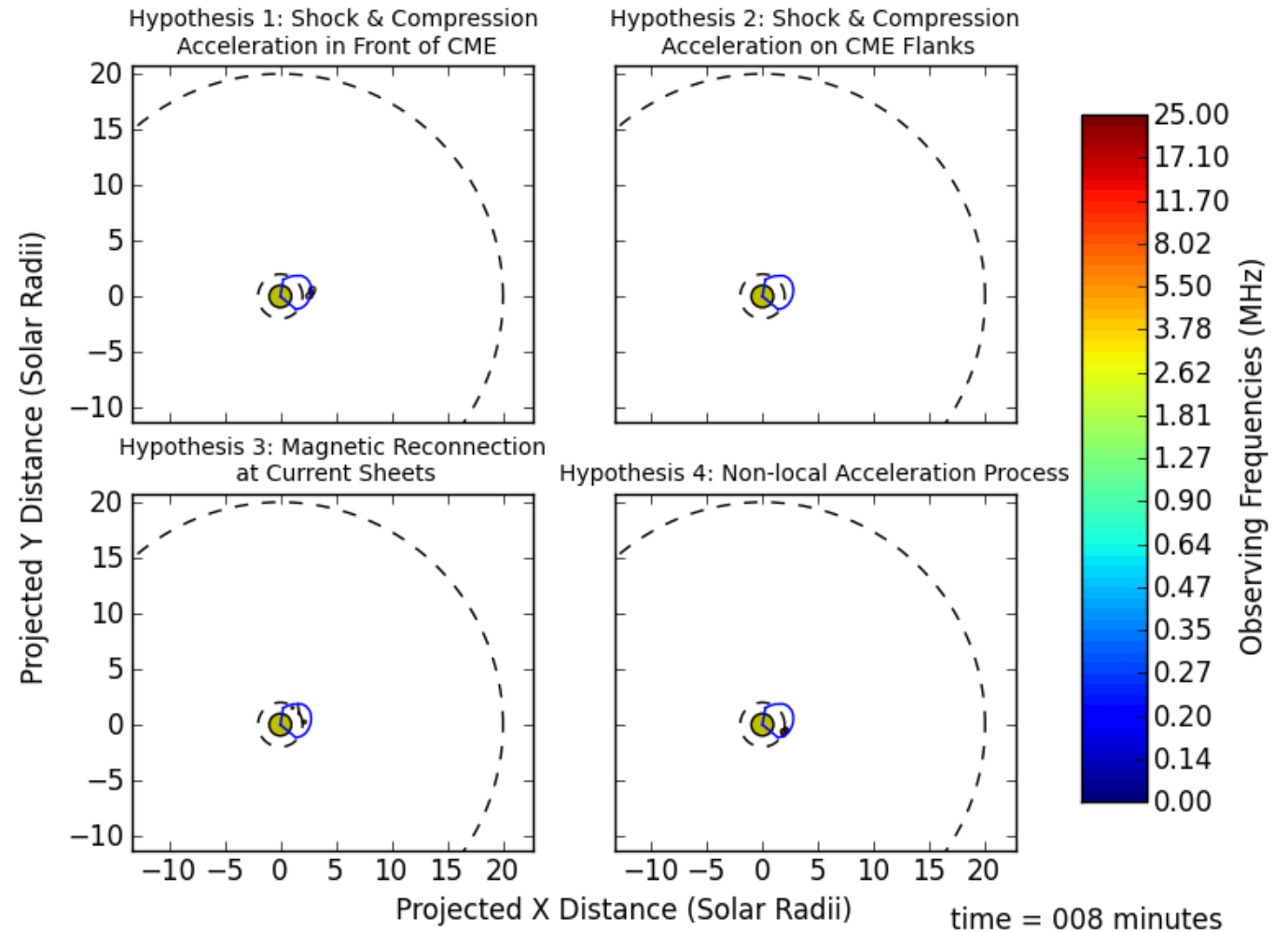
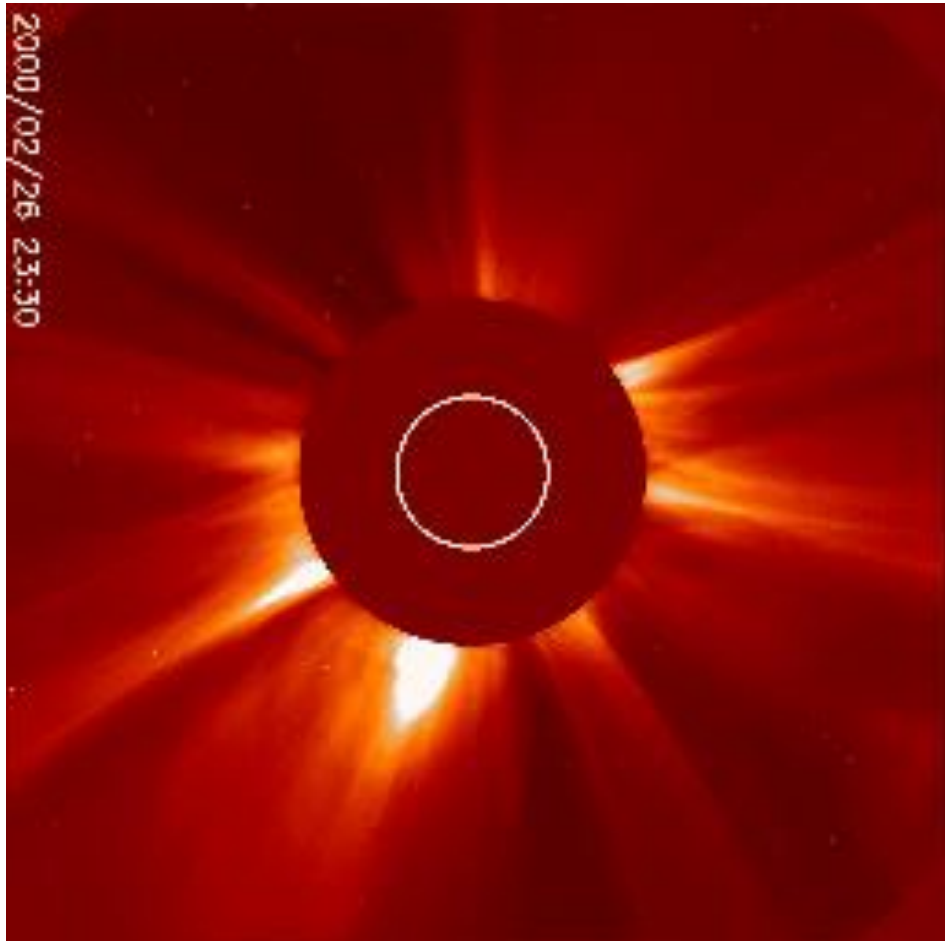
Launch is about here
And we operate for a year

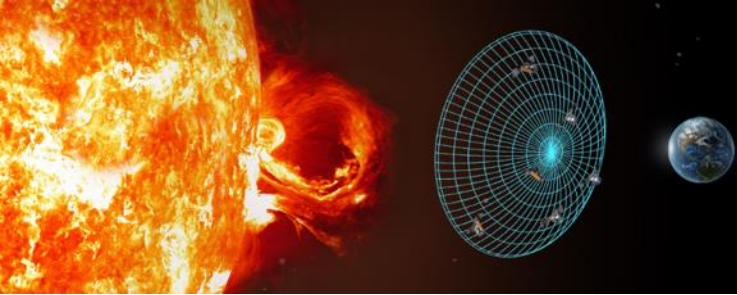
<https://www.swpc.noaa.gov/products/solar-cycle-progression>





Determining the Particle Acceleration Sites of CMEs

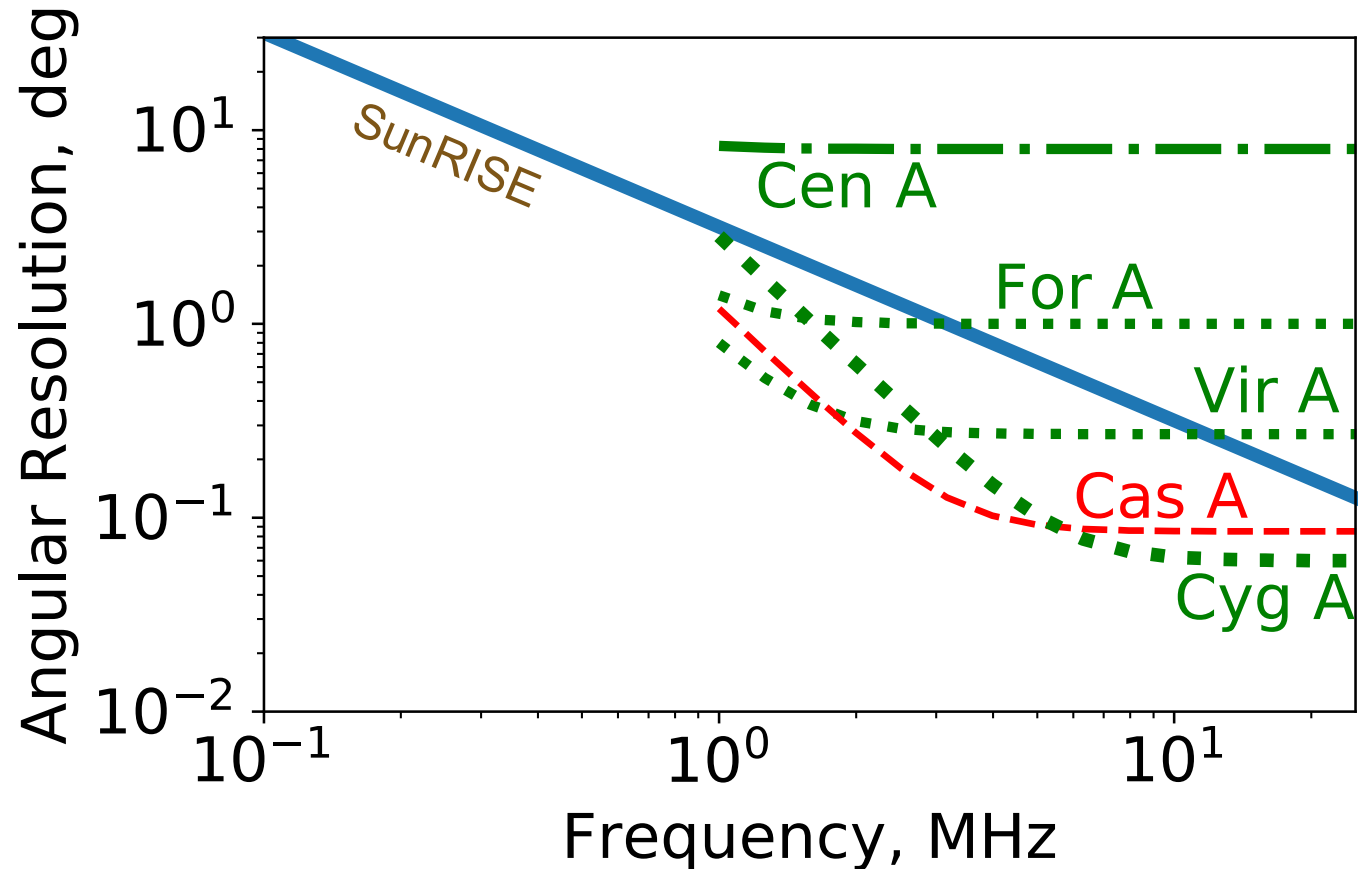


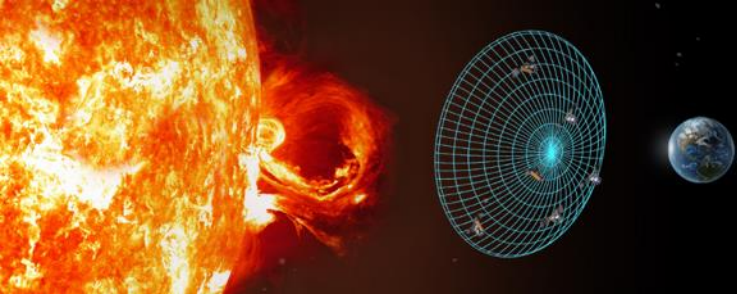


Low-Frequency Radio Astronomy



- SunRISE angular resolution tuned for localization of solar radio bursts.
- Examples of easily detectable sources are shown for comparison (there may be other sources as well)
- Galactic free-free absorption becomes an issue for low Galactic latitude sources (e.g., Cas A and Cyg A) below about ~1 MHz.





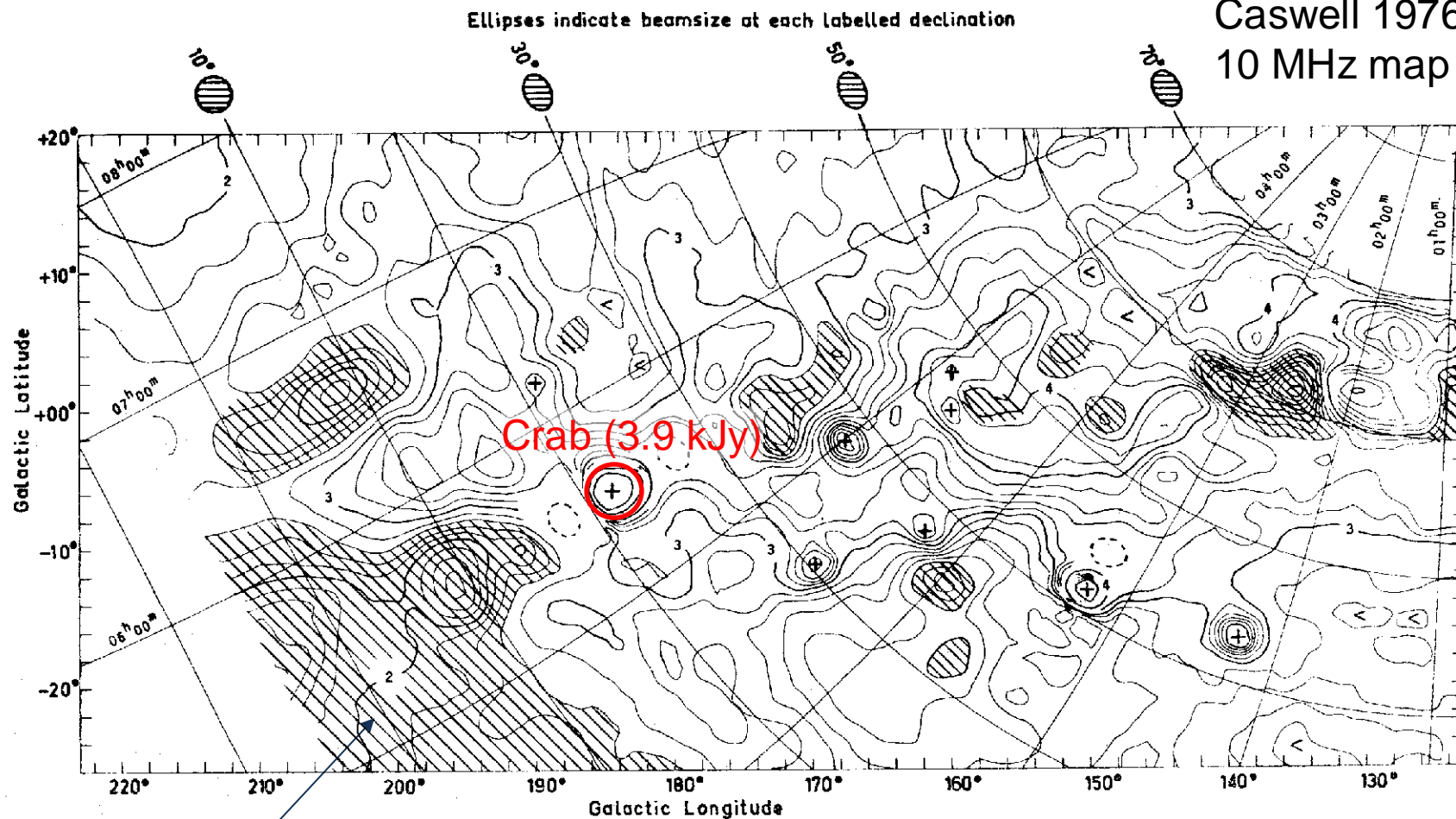
All-Sky Maps



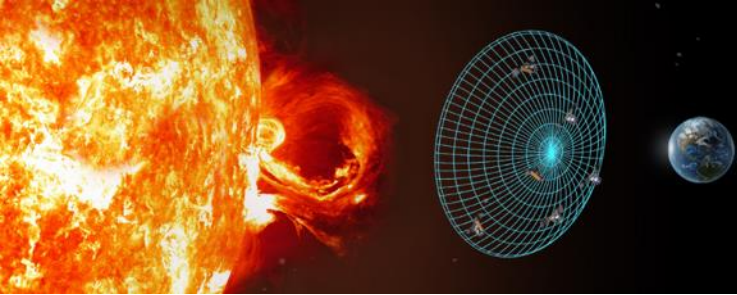
SUNRISE
SUN RADIO INTERFEROMETER SPACE EXPERIMENT

SunRISE could potentially produce all-sky maps with ~order of magnitude improvement in angular resolution from previous maps of Caswell 1976 (10 MHz) and Cane & Whitman 1976 (3.7 – 16.5 MHz).

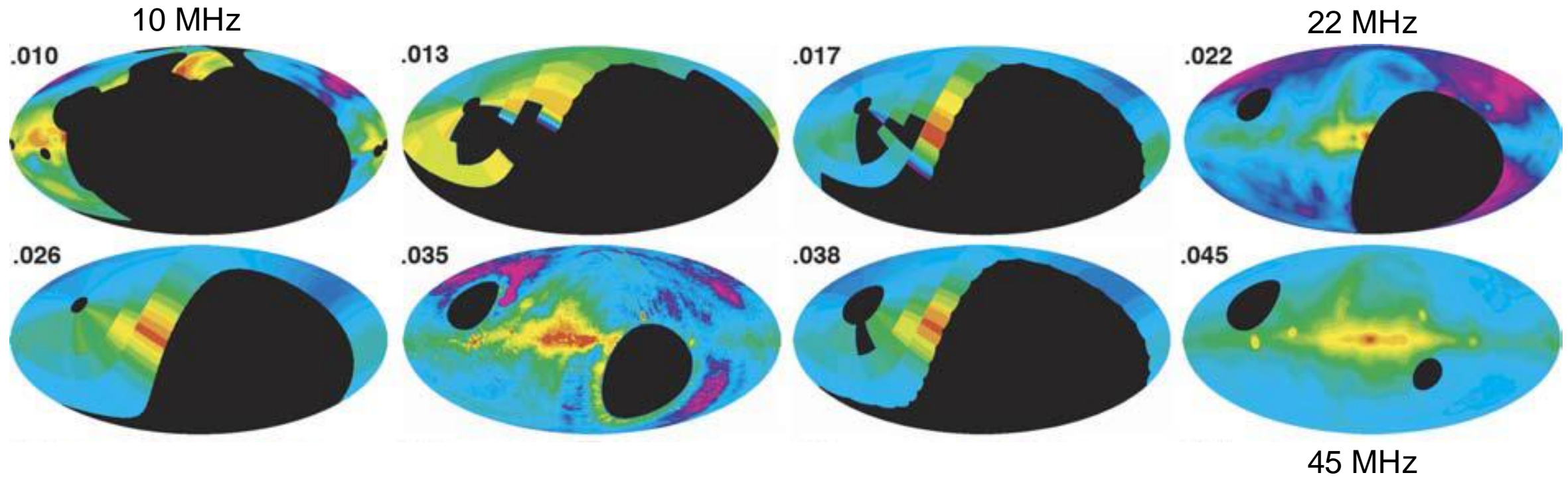
Caswell 1976,
10 MHz map



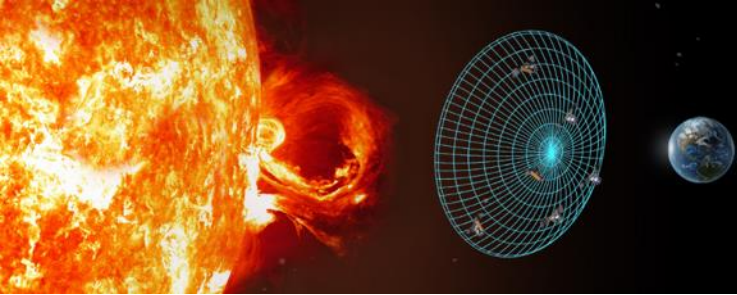
Hatched areas indicate absorption features corresponding to large-diameter H II regions.
20,000 K contours, contour labels in units of 100,000 K.



Low-Frequency Sky Maps (Ground-Based)



de Oliveira-Costa+ (2008)

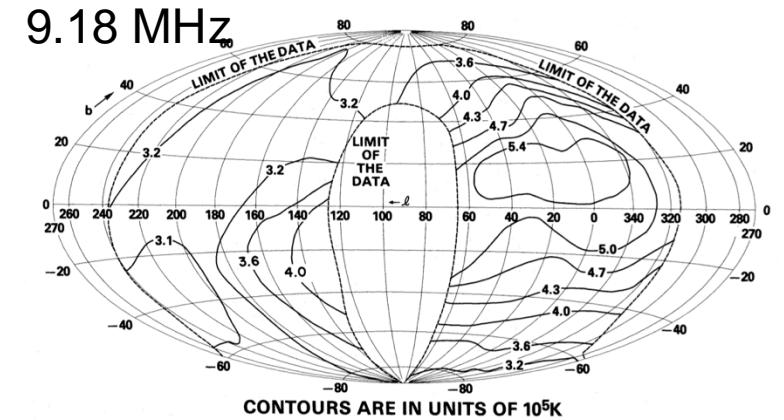
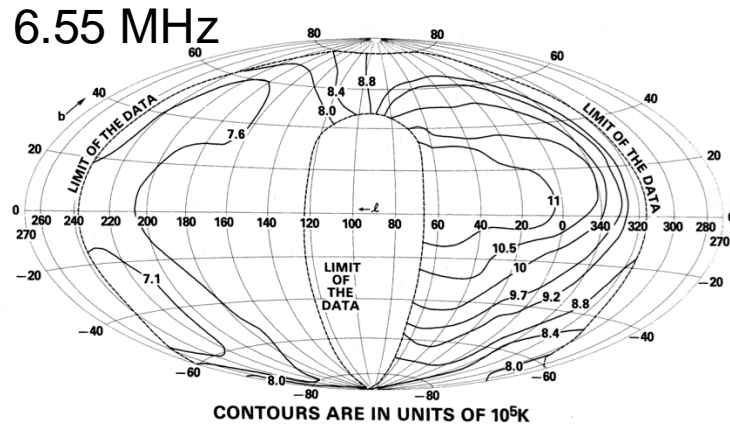
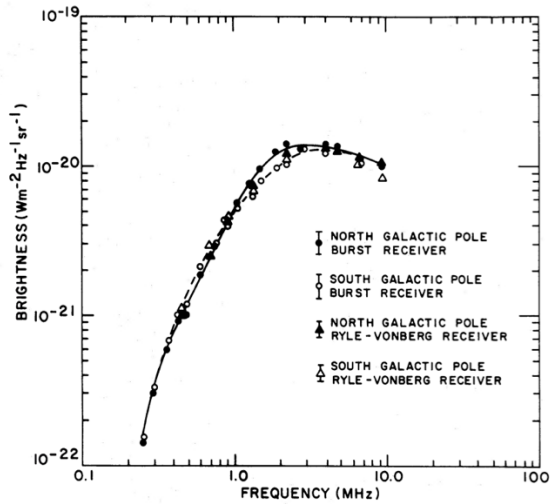
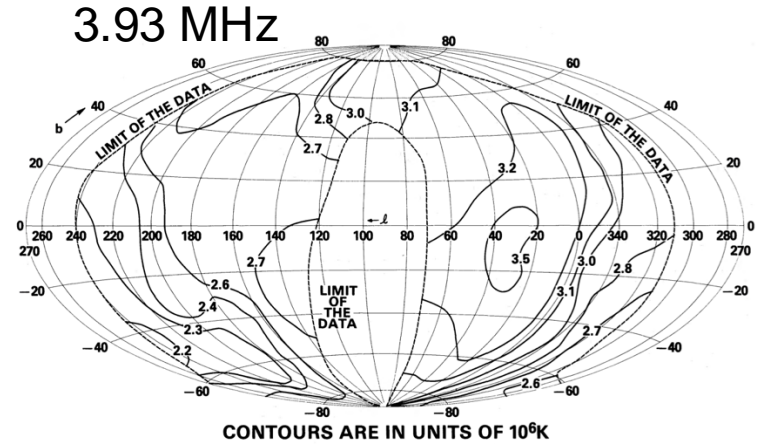
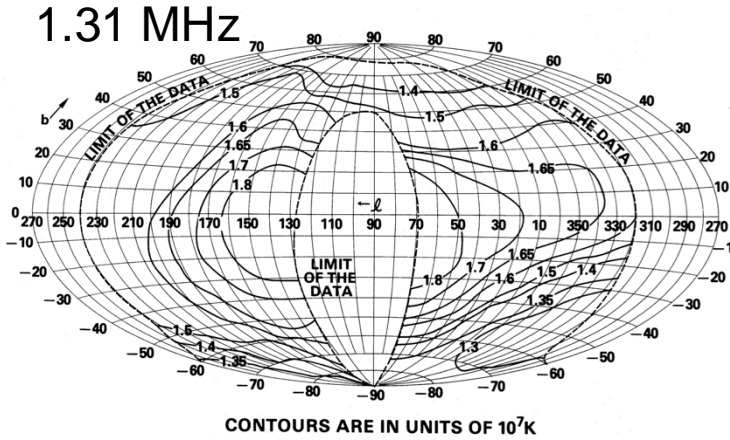
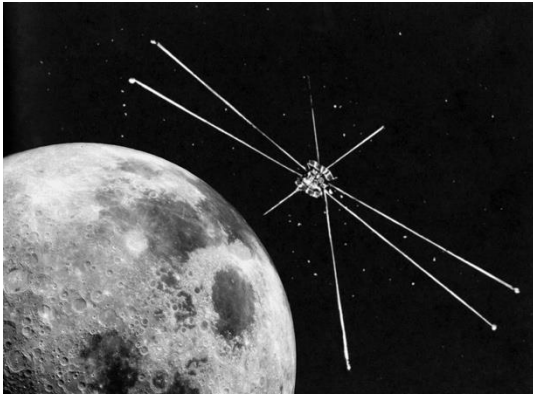


Low-Frequency Sky Maps (Space-Based)

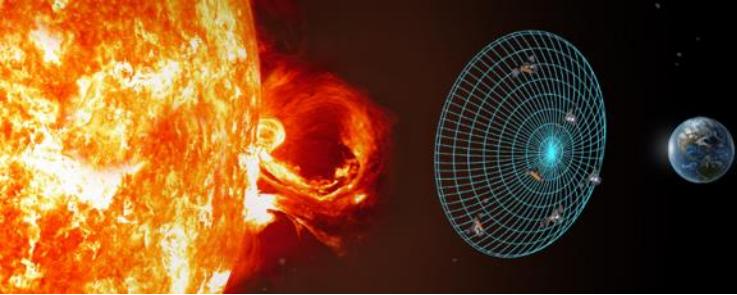


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Radio Astronomy Explorer-2 Mission 1973-1977



Novaco & Brown (1978)



Spatial Resolution

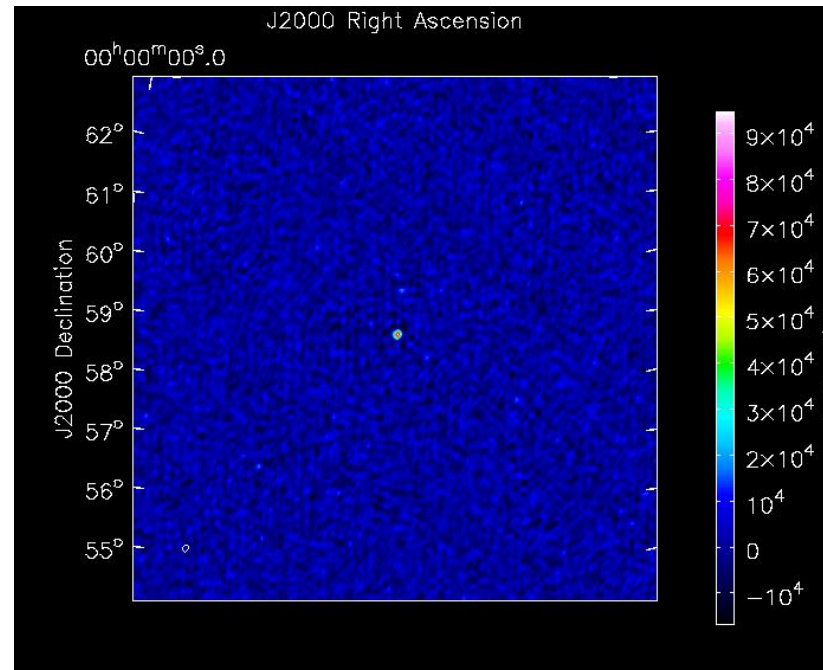


SUNRISE

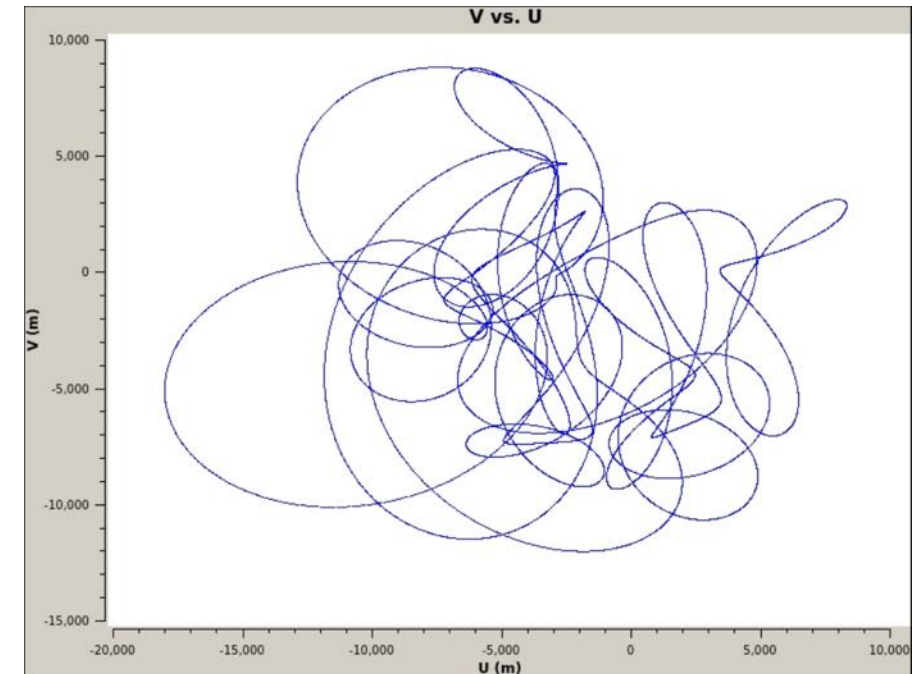
SUN RADIO INTERFEROMETER SPACE EXPERIMENT

UV plane coverage sufficient for imaging bright point-like sources.

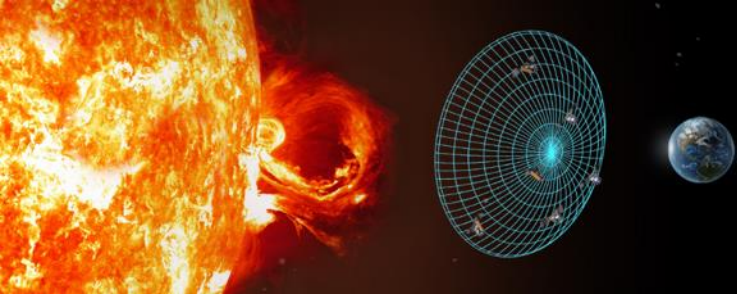
Noisy Cas A Image (10 MHz)



UV Coverage

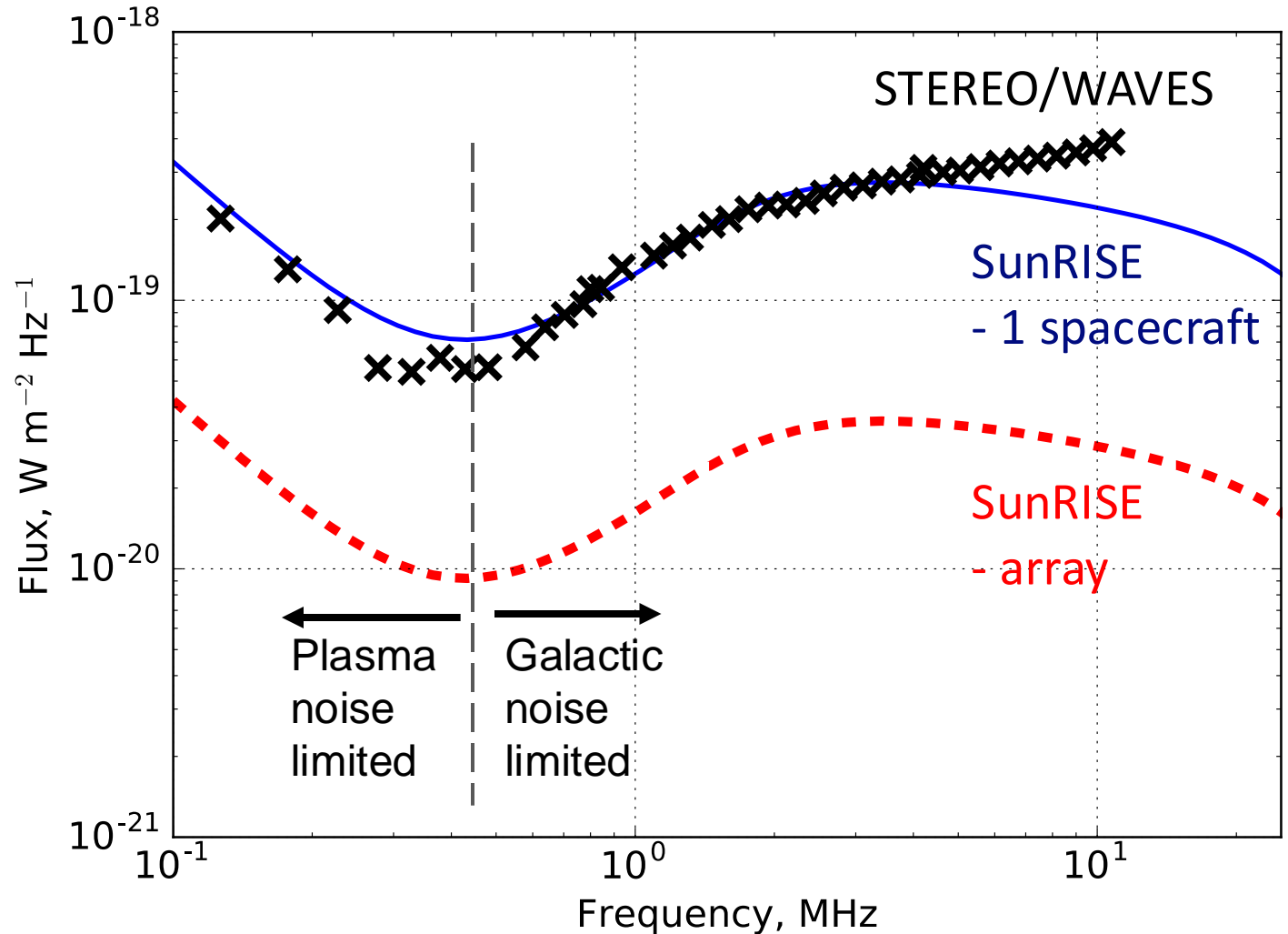


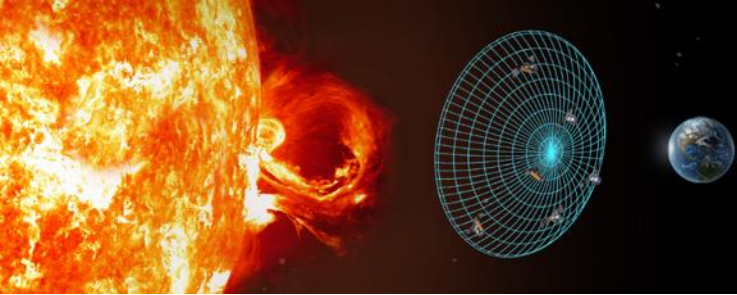
Repeats every 25 days.



Sensitivity

- Each “snapshot” (0.66 ms integration time, 6.1 kHz bandwidth) is background noise limited.
- SunRISE sensitivity is comparable to Wind/WAVES and the STEREO/WAVES receivers.
- Array: 6 spacecraft, 2 polarizations improves the sensitivity by a factor of 8.5.





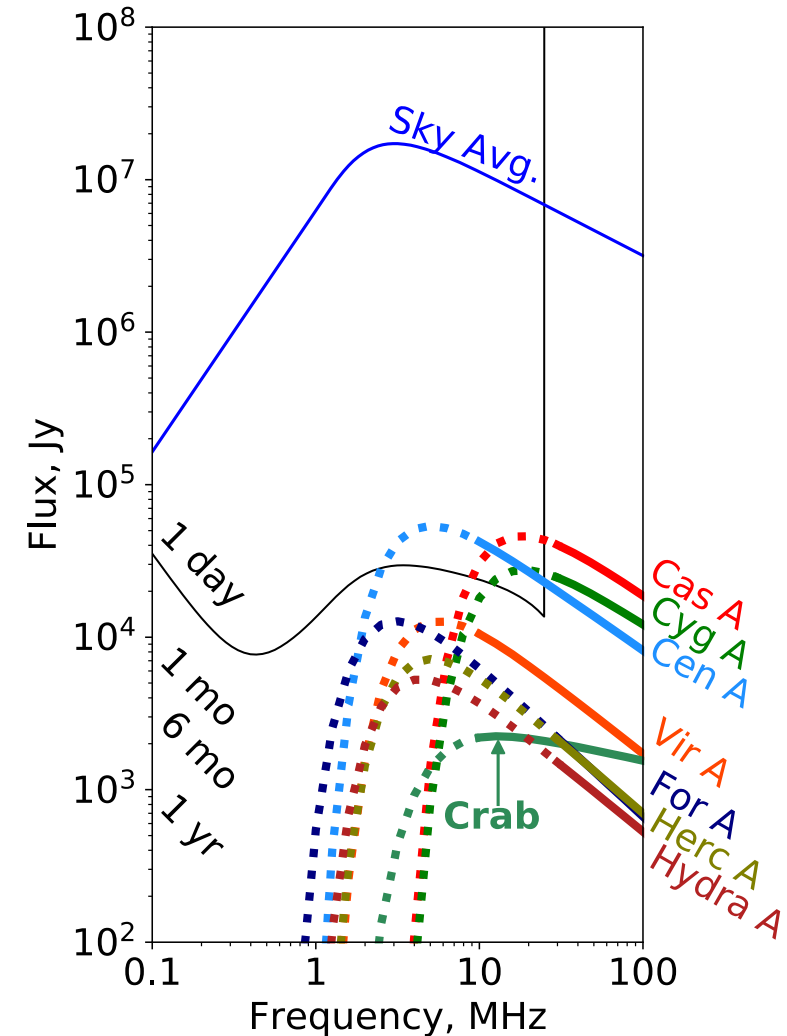
Sensitivity

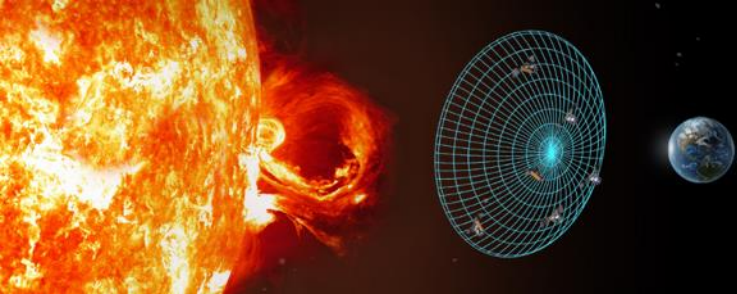


SUNRISE

SUN RADIO INTERFEROMETER SPACE EXPERIMENT

- SunRISE integrated sensitivity would be capable of imaging the A-team sources at frequencies below the ionospheric cutoff.
- Cas A and Cyg A expected to be absorbed at frequencies <10 MHz due to low Galactic latitudes.



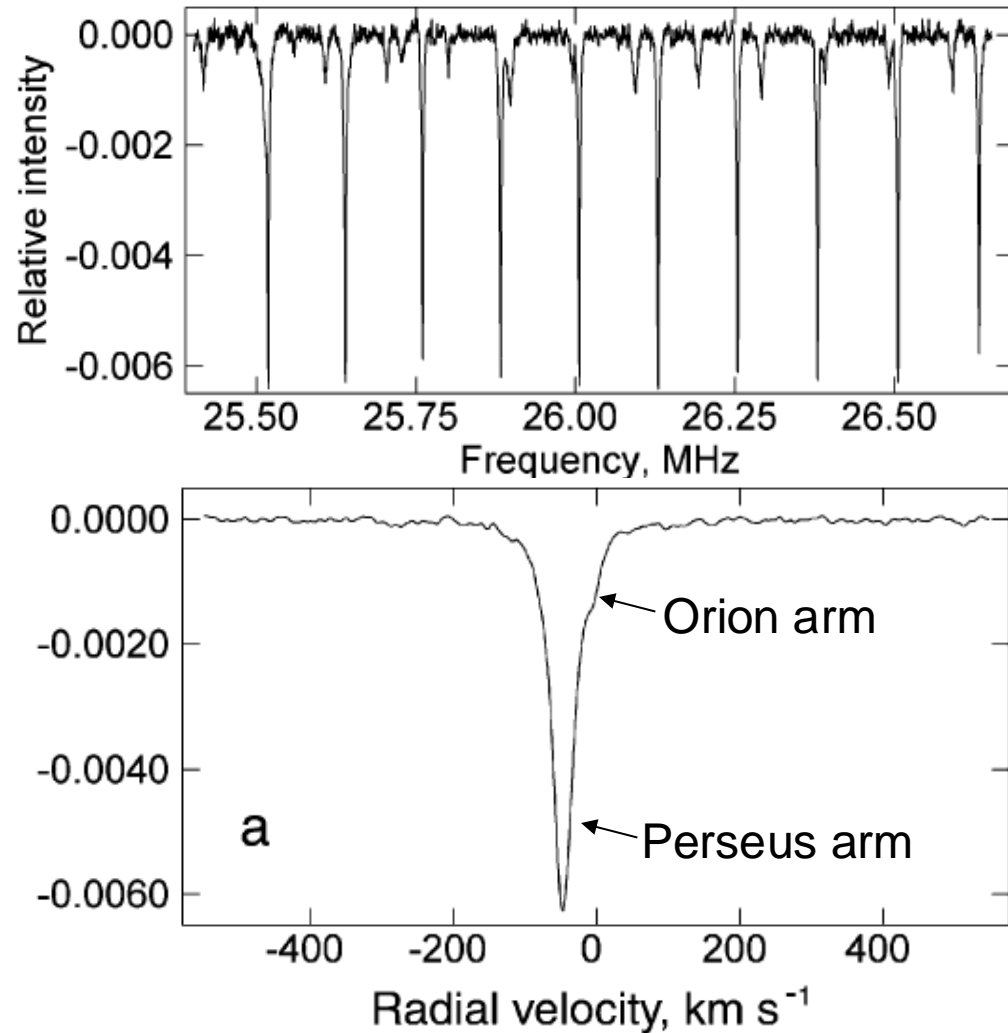


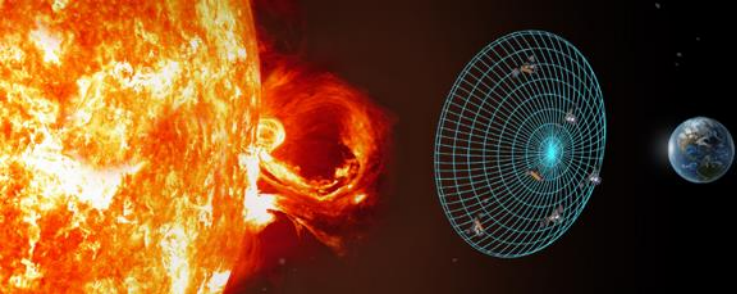
Radio Recombination Lines



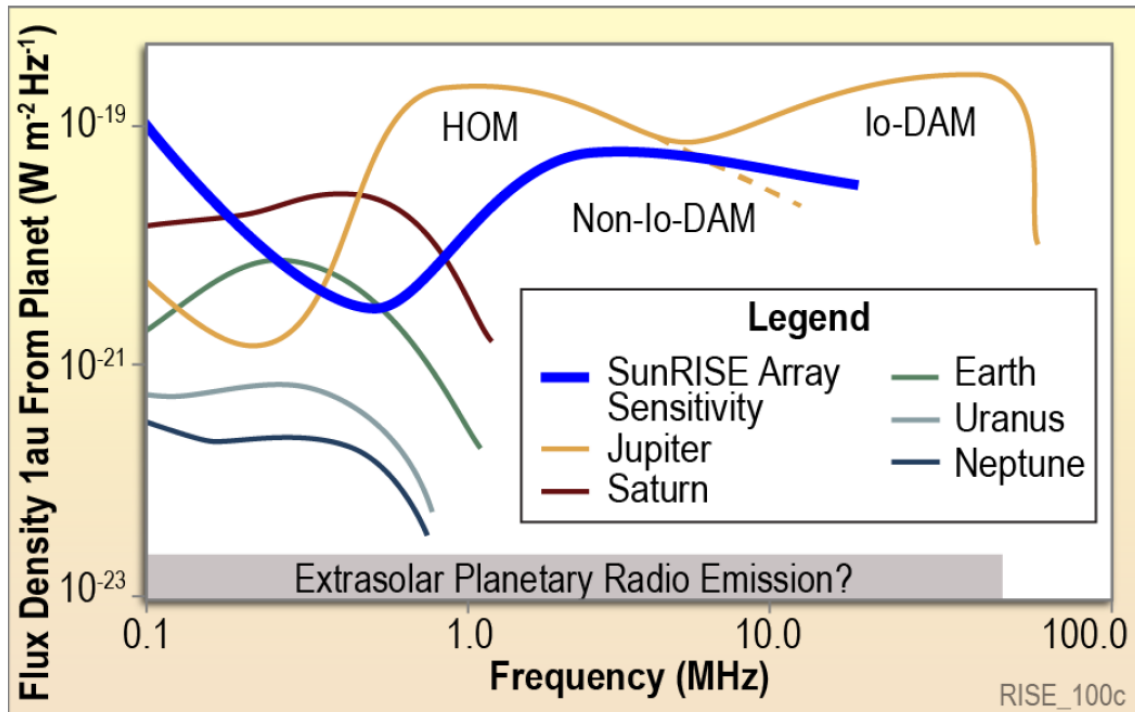
SUNRISE
SUN RADIO INTERFEROMETER SPACE EXPERIMENT

- SunRISE would search for radio recombination lines at frequencies below 25 MHz.
- Using Cas A as an illumination source and stacking the comb structure of radio recombination lines could enable a 5σ detection of line depths of $\gtrsim 2.5 \times 10^{-3}$ or greater.

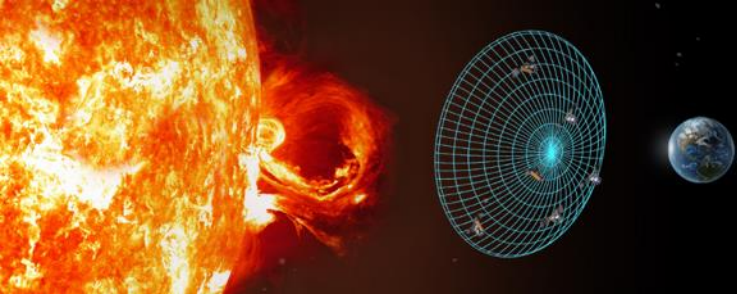




Planetary Science



- Typical planetary radio emission seen from 1 au distance compared to SunRISE sensitivity in 1 second, 1 Hz measurement. Ten minute, 2 MHz integrations would be up to 10,000x more sensitive
- Jupiter should be detectable most of the time
- Saturn, Uranus, and Neptune are a stretch but would be very exciting if detected

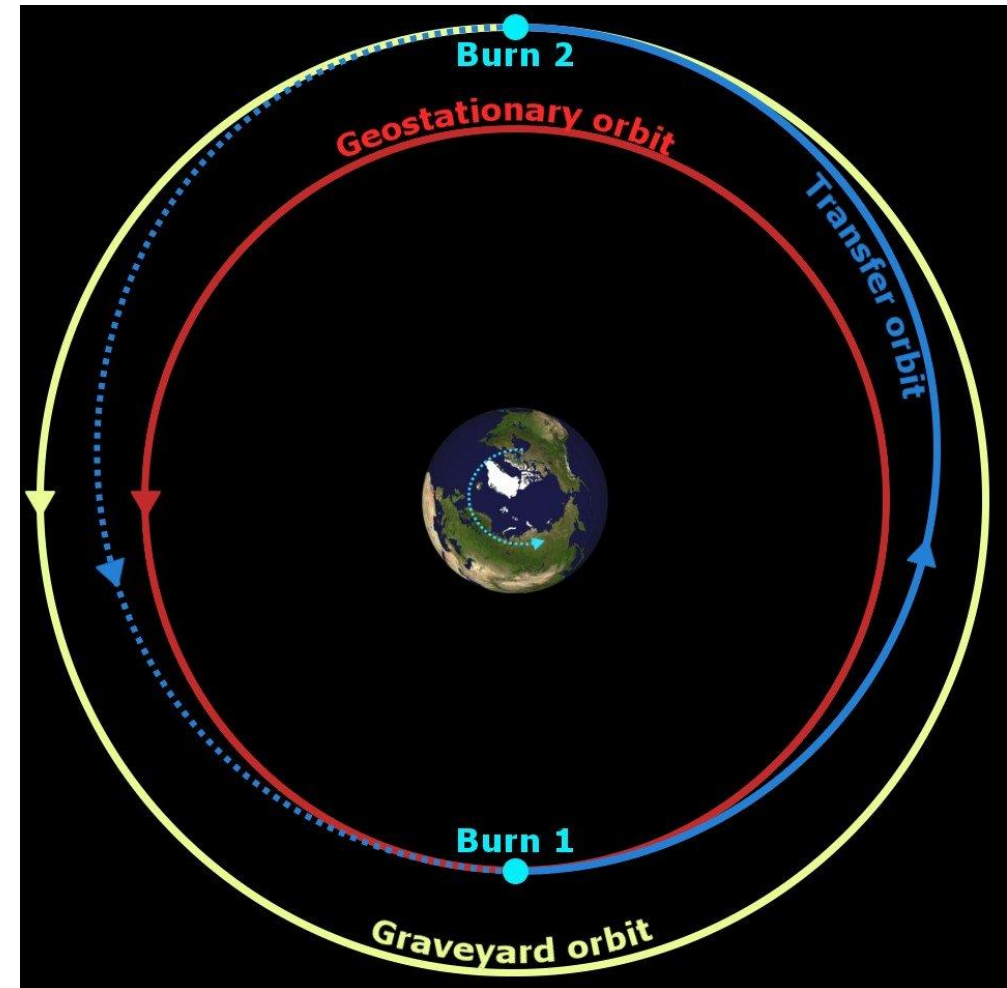


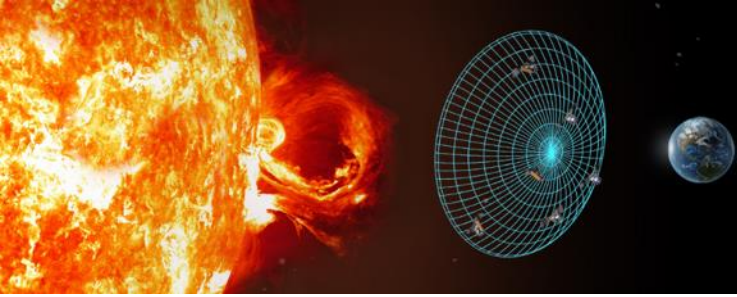
Observing with Ground-Based Radio Telescopes



SUNRISE
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- SunRISE is in a ~37,000 km altitude Supersynchronous orbit (a.k.a. GEO graveyard orbit).
- 25 hour period means SunRISE shifts 0.6° / hour with respect to a point on the ground.
- Potential for synchronized observations with ground based telescopes.
- Data type is an amplitude and phase with 6.1 kHz bandwidth acquired every 100 ms.





Summary



- SunRISE is in a ~37,000 km altitude Supersynchronous orbit (a.k.a. GEO graveyard orbit).
- 25 hour period means SunRISE shifts 0.6° / hour with respect to a point on the ground.
- Potential for synchronized observations with ground based telescopes.
- Data type is an amplitude and phase with 6.1 kHz bandwidth acquired every 100 ms.



SUNRISE

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